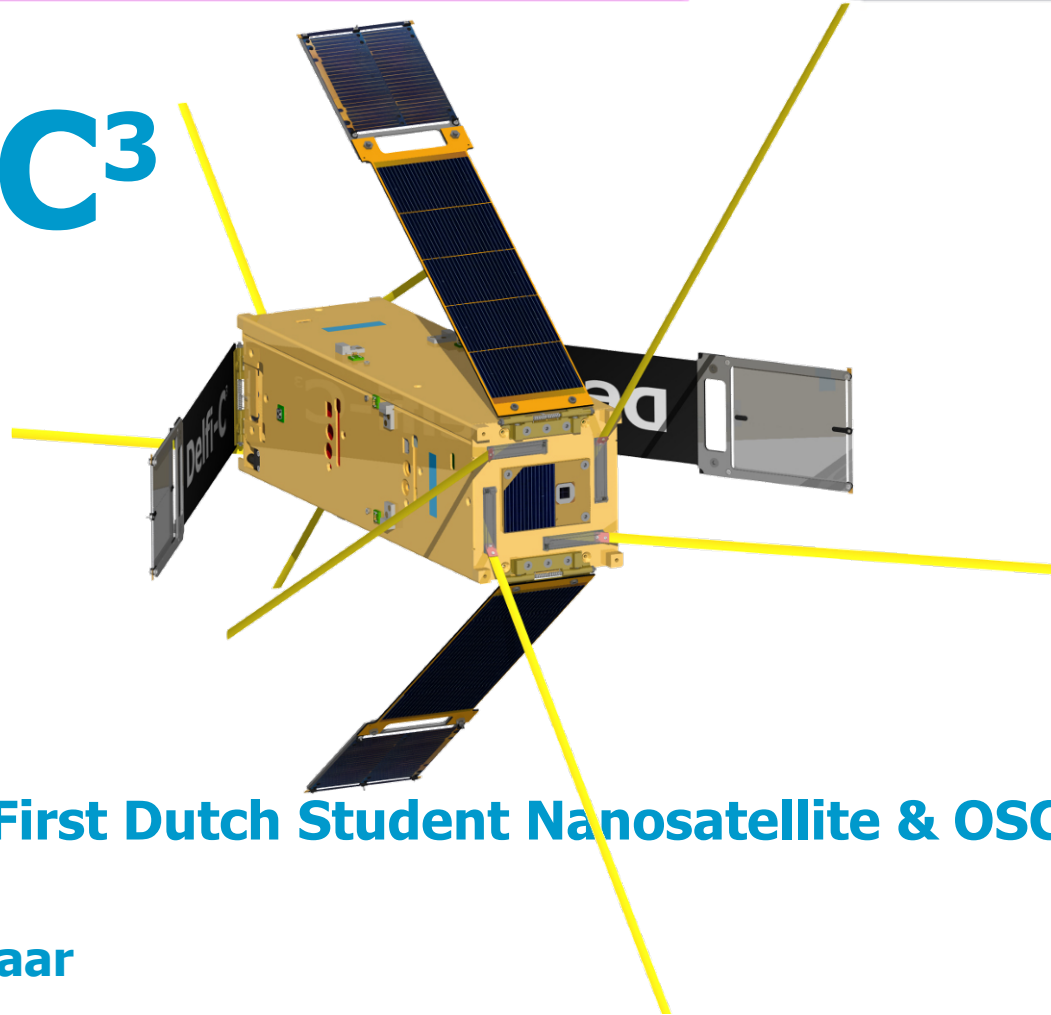


Delfi-C³



Realizing the First Dutch Student Nanosatellite & OSCAR

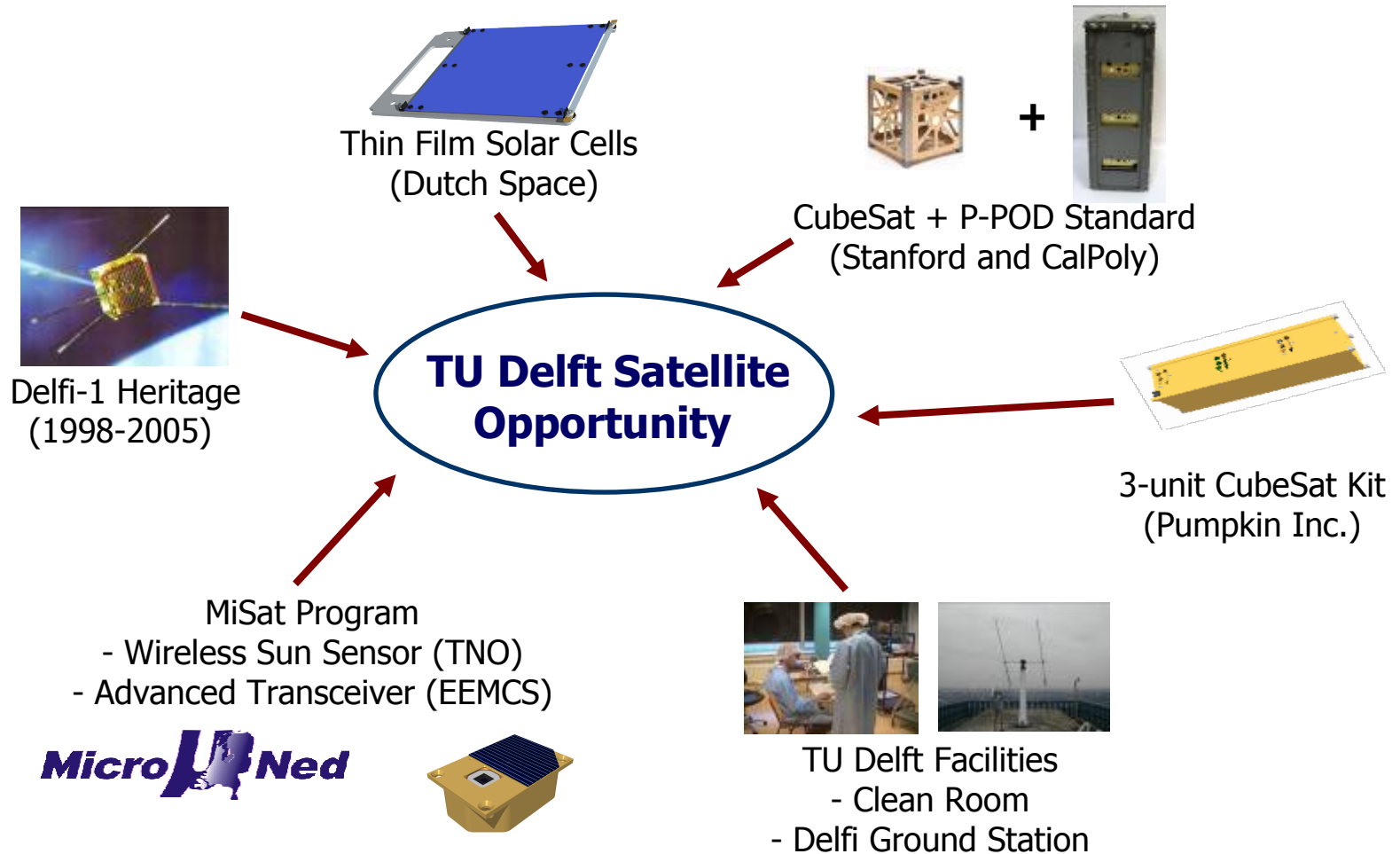
Wouter Weggelaar

PA3WEG

Delft University of Technology, The Netherlands

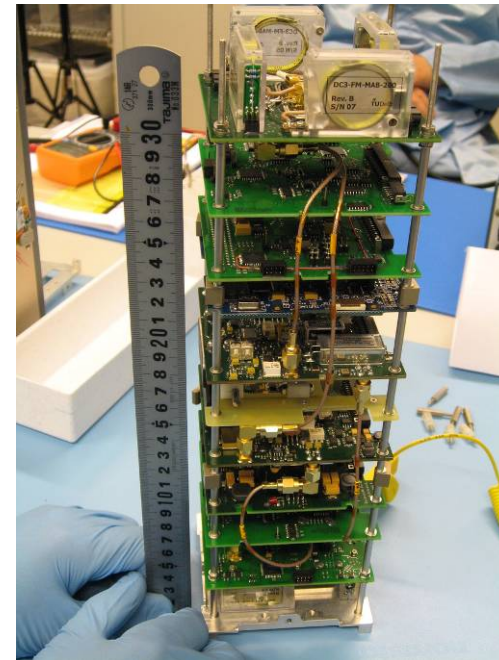
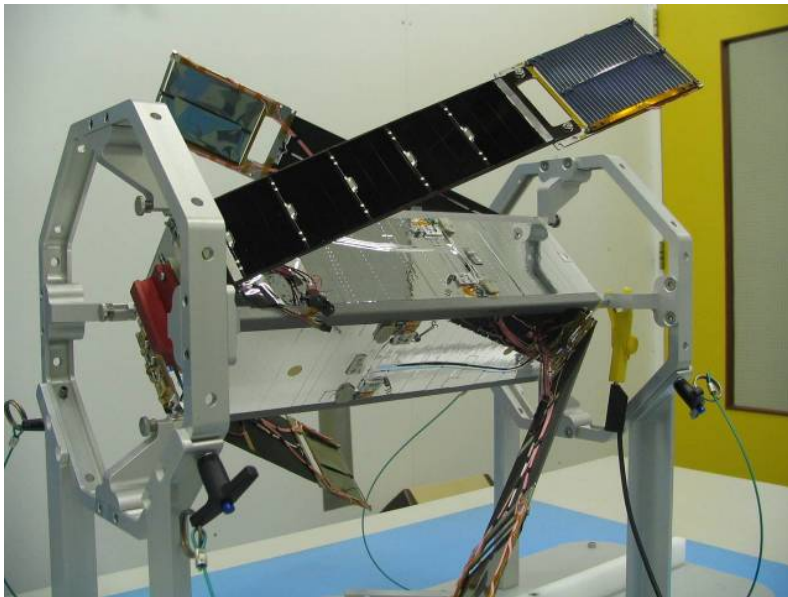


November 2004:

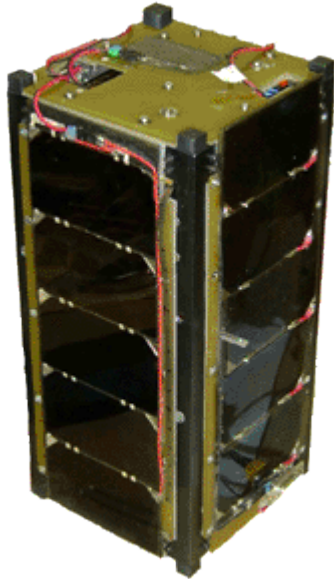


April 2008:

- Satellite ready for launch
- Over 60 students have worked on Delfi-C3 (+ 3 new HAMs!)
- Successor Delfi-n3Xt



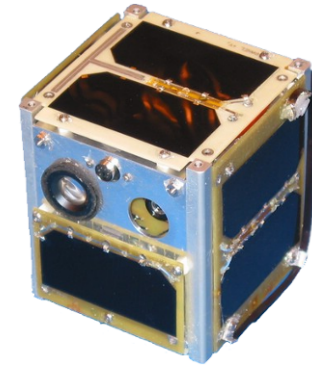
CubeSats



U. Of Illinois ION



Cal Poly CP1



CAN-X1

Delfi-C³ Mission Overview - Objectives

Summarized Technical Objectives:

- Perform in-orbit test of a Thin Film Solar Cells
- Perform in-orbit test of an Autonomous Wireless Sun Sensor
- Create a distributed ground station network for Delfi-C3 and future missions

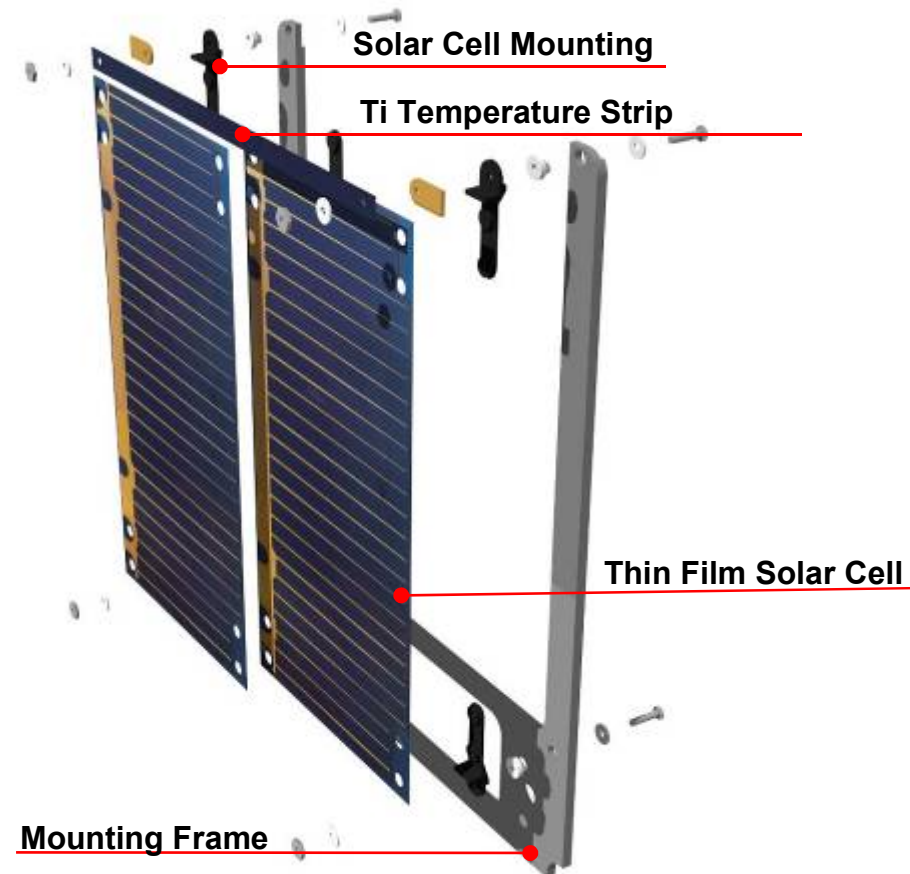
Summarized Educational Objectives:

- Provide interdisciplinary hands-on engineering experience
- Develop teamwork, leadership, and communication skills
- Interface with the MSc. programs of TU Delft
- Provide an opportunity to a variety of educational organizations to participate



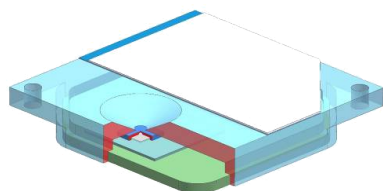
Thin Film Solar Cells (Dutch Space)

- First flight opportunity
- Innovative technology:
 - Thin film titanium substrate $\sim 25 \mu\text{m}$
 - High power to mass ratio
 - Very low stack height
- IV-curve measurement
- Temperature measurement
- Modular payload
- *No body mounting*

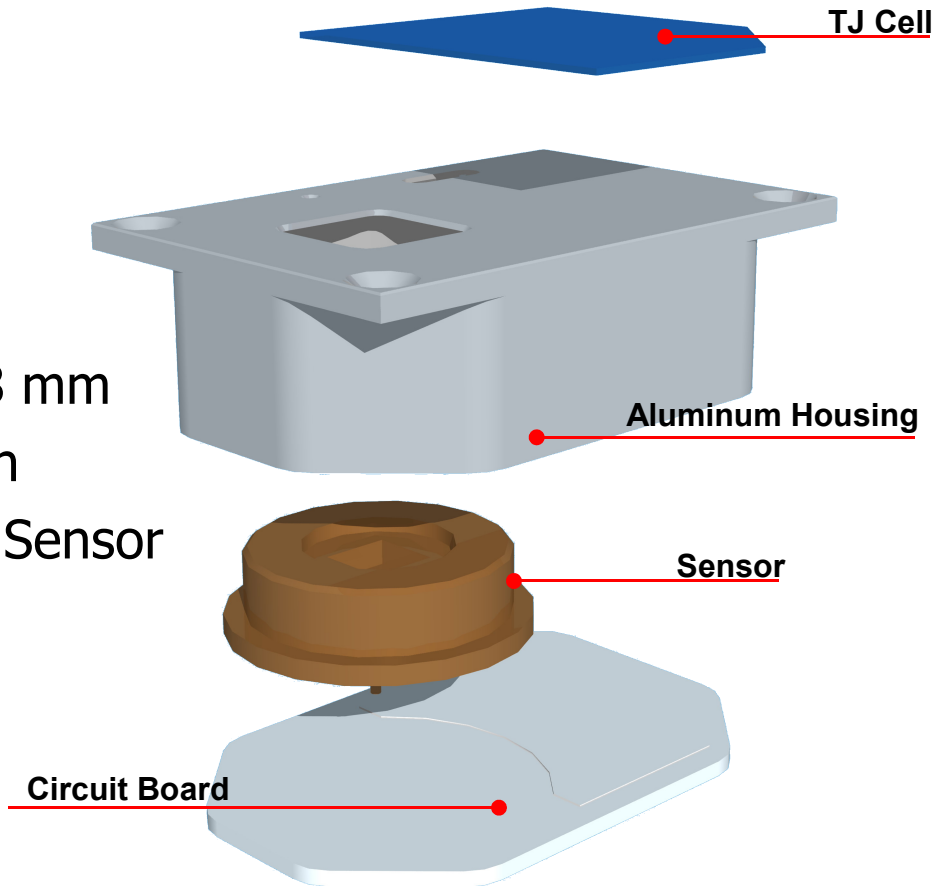
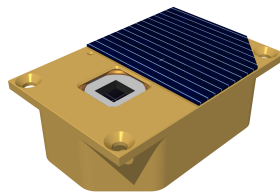


Autonomous Wireless Sun Sensor (TNO)

- Analog Quadrant Sun Sensor (OTS)
- Wireless RF-Interface
 - UHF Link (915 MHz)
 - Patch antenna on sensor
 - 1 RF-receiver connected to OBC
- Integrated GaAs solar cell
- Sensor envelope $\sim 60 \times 40.5 \times 17.8$ mm
- 2 Sensor units, mass ~ 75 g each
- Predecessor to Micro Digital Sun Sensor



Digital Sun Sensor [TNO]



Mission Characteristics, launch and Realization

- Design and development by a self-organized student team
- Telemetry gathering through Radio Amateurs
- *Mode U/V Linear transponder*
- Designed for 1 year LEO



X-POD



PSLV

- “Piggyback” Launch in X-POD NLS-4 launch Polar Satellite Launch Vehicle (PSLV), India, **21st of April 2008**

Orbit:

- Inclination: 97.91°
- Sun synchronous
- Altitude: 630 km, circular orbit

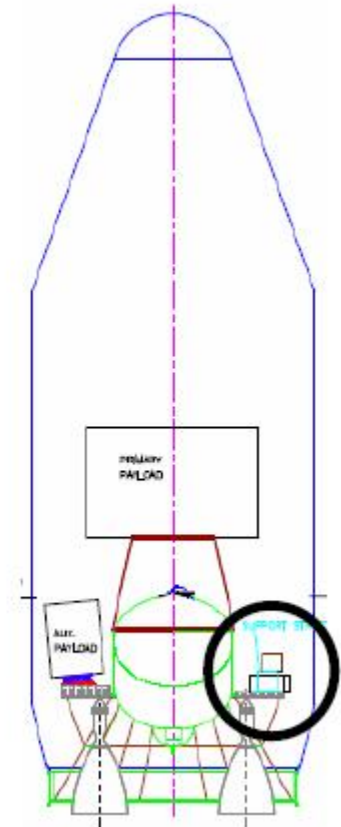
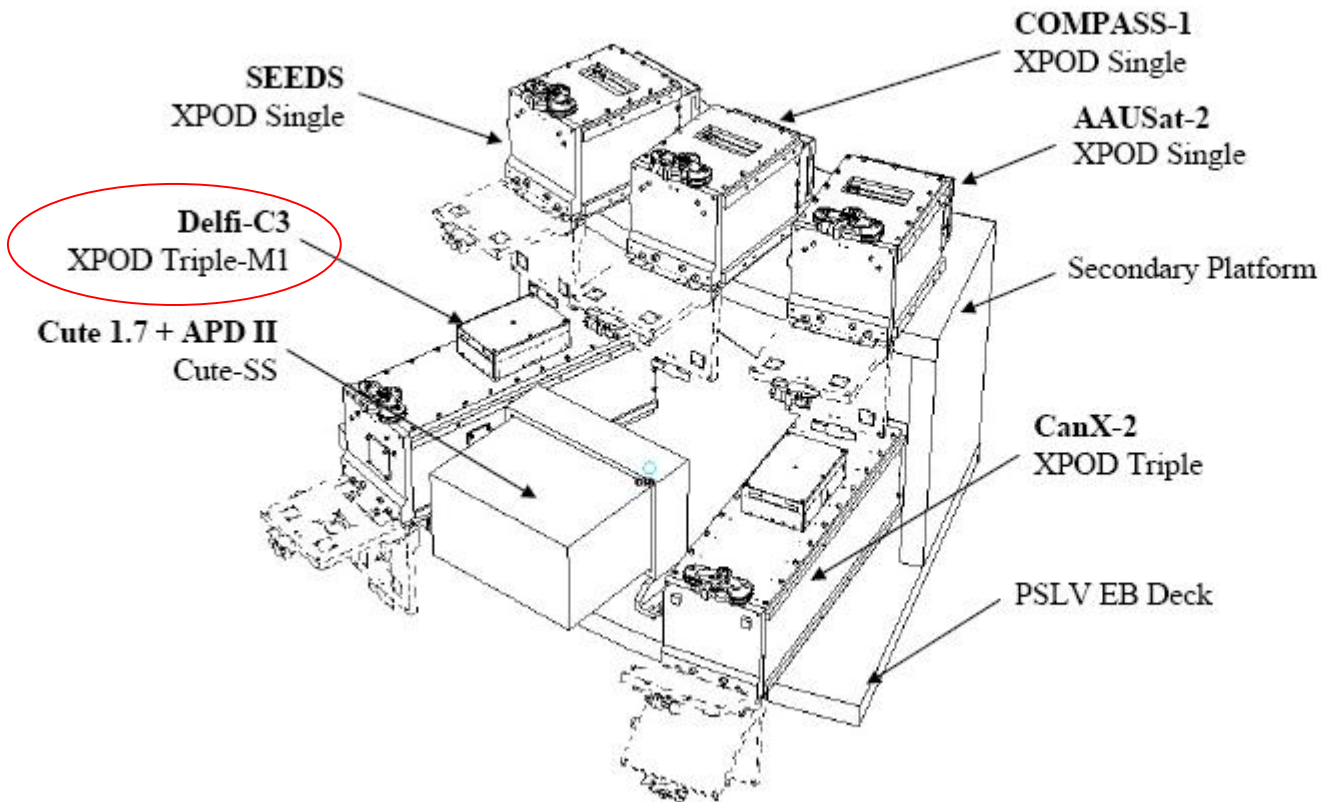
- 3 months science mission, after which linear transponder mode



Shriharikota Launch Site



Launch configuration



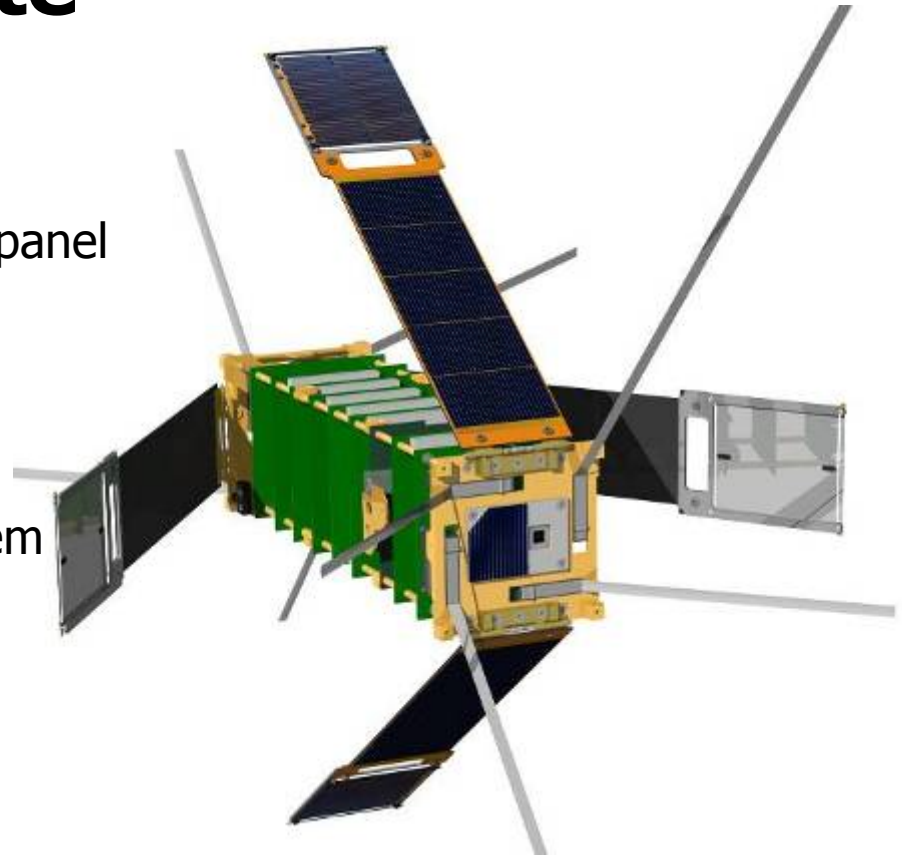
The Delfi-C3 satellite

- 3 unit CubeSat structure, 3kg
- 2 AWSS payload units
- 4 deployable panels at 35 degrees (max/min power):
 - Carbon Fiber Reinforced Plastic
 - TFSC payload suspension frame
 - 5 TEC1 GaAs TJ solar cells
- 2.5 W min. power available
- 8 antennas:
 - 4 VHF 50 cm downlink
 - 4 UHF 18 cm uplink
- ***No battery***
- ***No active attitude control***

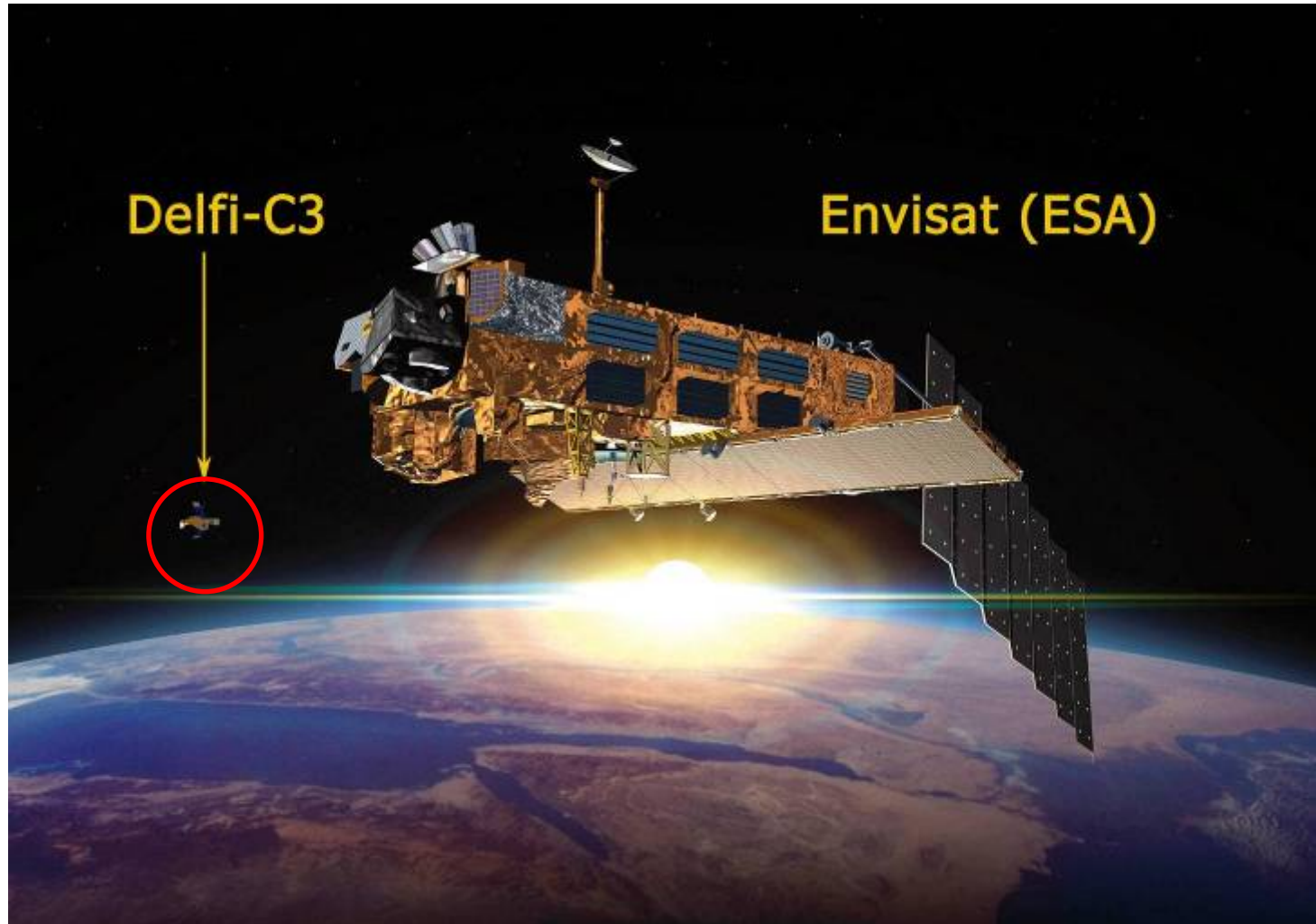


The Delfi-C3 satellite

- EPS
 - 1 DC DC converter per solar panel
 - Current measurement
- CDHS
 - TI MSP430 OBC
 - Microchip PIC18LF4220 microcontrollers per subsystem
- Attitude Control
 - Magnetic hysteresis rods
- COMMS
 - Two Redundant Transceivers
- Standard board interface
 - I²C bus
 - 12 V DC power bus
- Passive thermal subsystem (thermal tapes)

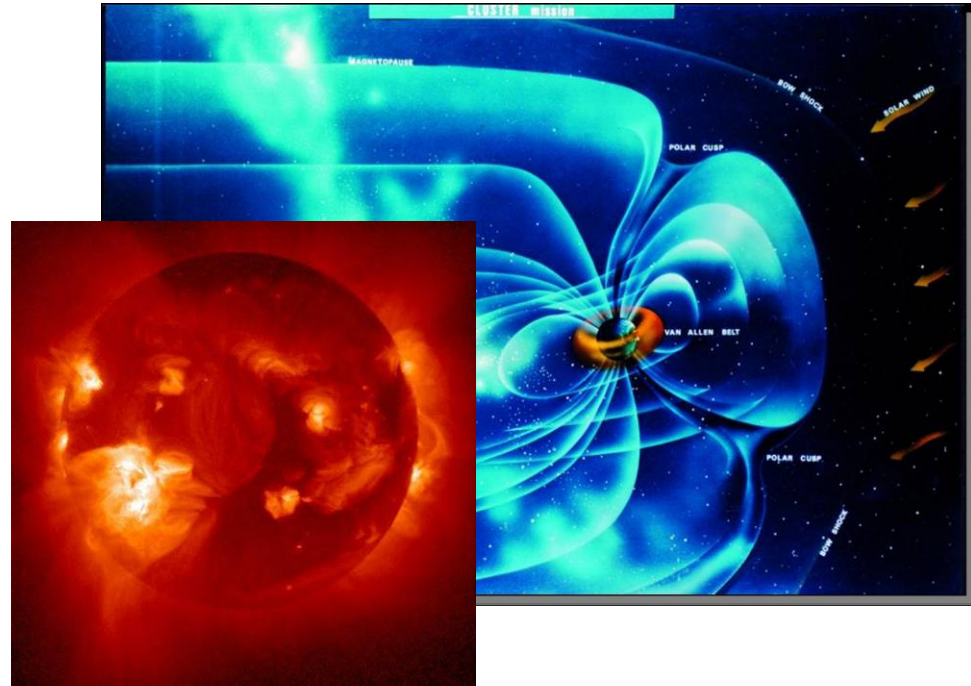


Delfi-C3 is Small...

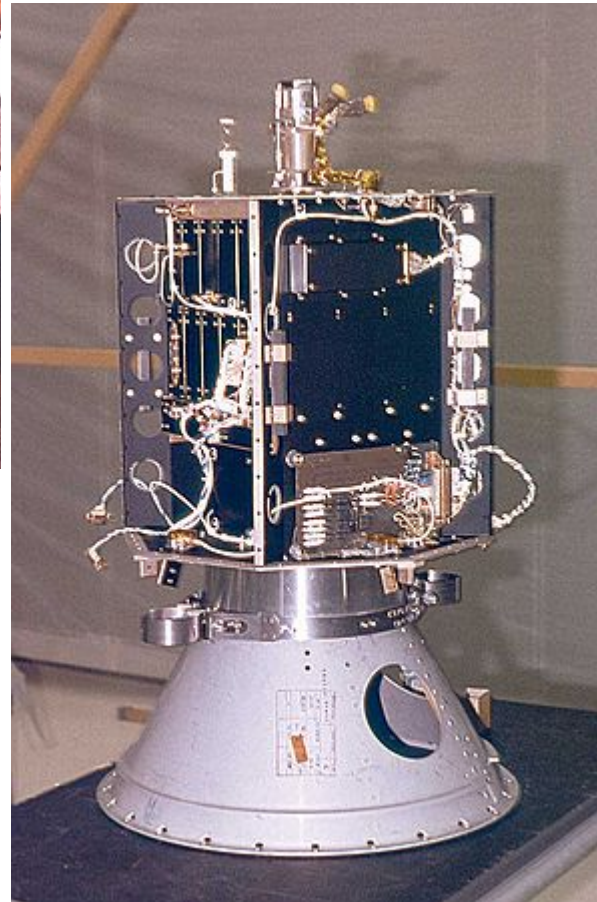
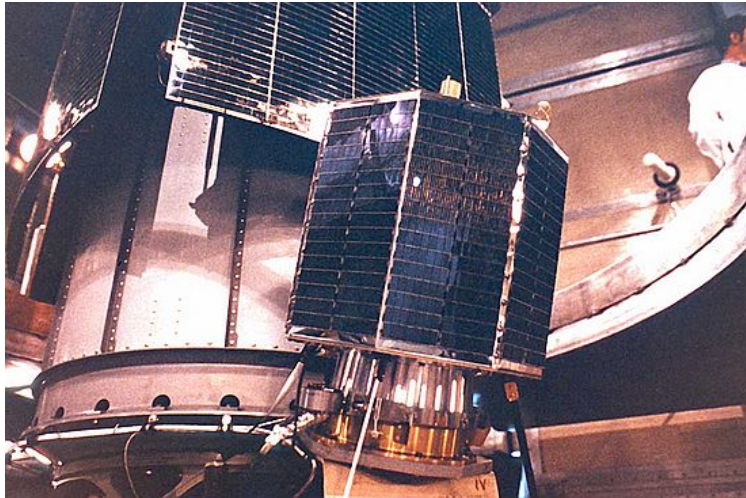


Designing electronics for space

- Radiation
- Temperature
- Vacuum...
- EMC & ESD
- Radiation effects
- Low power / low voltage electronics
- Structural loads
- Thorough testing
- Redundancy
- Commercial Off The Shelf parts



AMSAT OSCAR-7 story...



Communications subsystem

- 2 redundant transceivers
- UHF Receiver
 - 70MHz first IF, 10.7MHz second IF, 455kHz 3rd IF (telecommand RX, based on MC3362DW)
 - Selectivity is more important than noise figure (+/-6dB NF)
 - SAW frontend filters
 - 3rd harmonic problem
 - Required in-band dynamic range is low
- Transponder IF based on AD8367 log amp + detector
- VHF transmitter
 - PIC AX.25 formatting, NRZ-I encoding, bit shaping (D/A)
 - 10.7MHz BPSK modulator (SA612), power combiner to combine with transponder IF
 - Conversion & amplification stage to 145MHz (Opamps)
 - MRF313 class A / AB final amplifier (400mW PEP) with ALC (QRP!!)



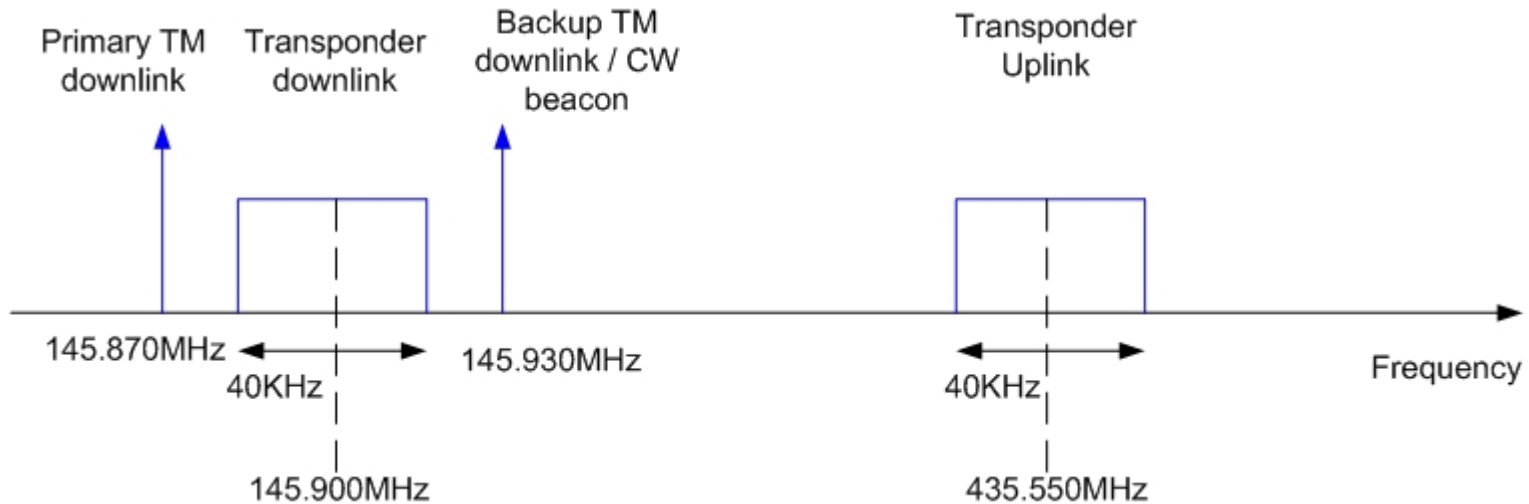
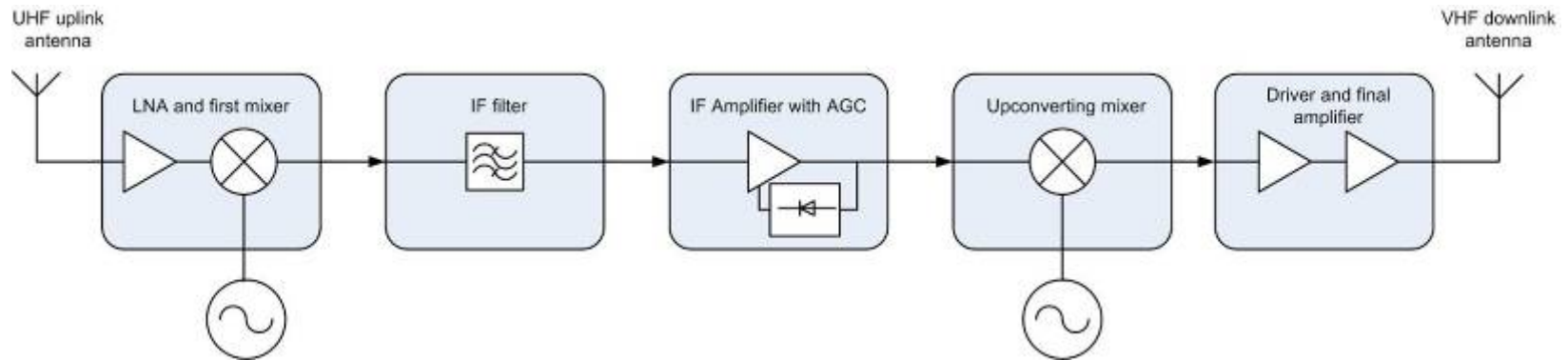
Frequencies

- Primary telemetry downlink: 145.870MHz
 - RC-BPSK, 1200bd AX.25, UI frames, 1 frame/sec
- Back-up: 145.930MHz
- Transponder downlink: 145.880-145.920MHz linear (inverting) + CW telemetry 40mW at 145.930MHz (*Hi Hi de Delfi-C3 Delfi-C3*)
- Transponder uplink: 435.570-435.530MHz
 - 40kHz passband, 400mW PEP
 - Simple transponder
 - No HELAPS
 - Basic AGC circuit
- Telecommand uplink: unpublished
 - Authentication

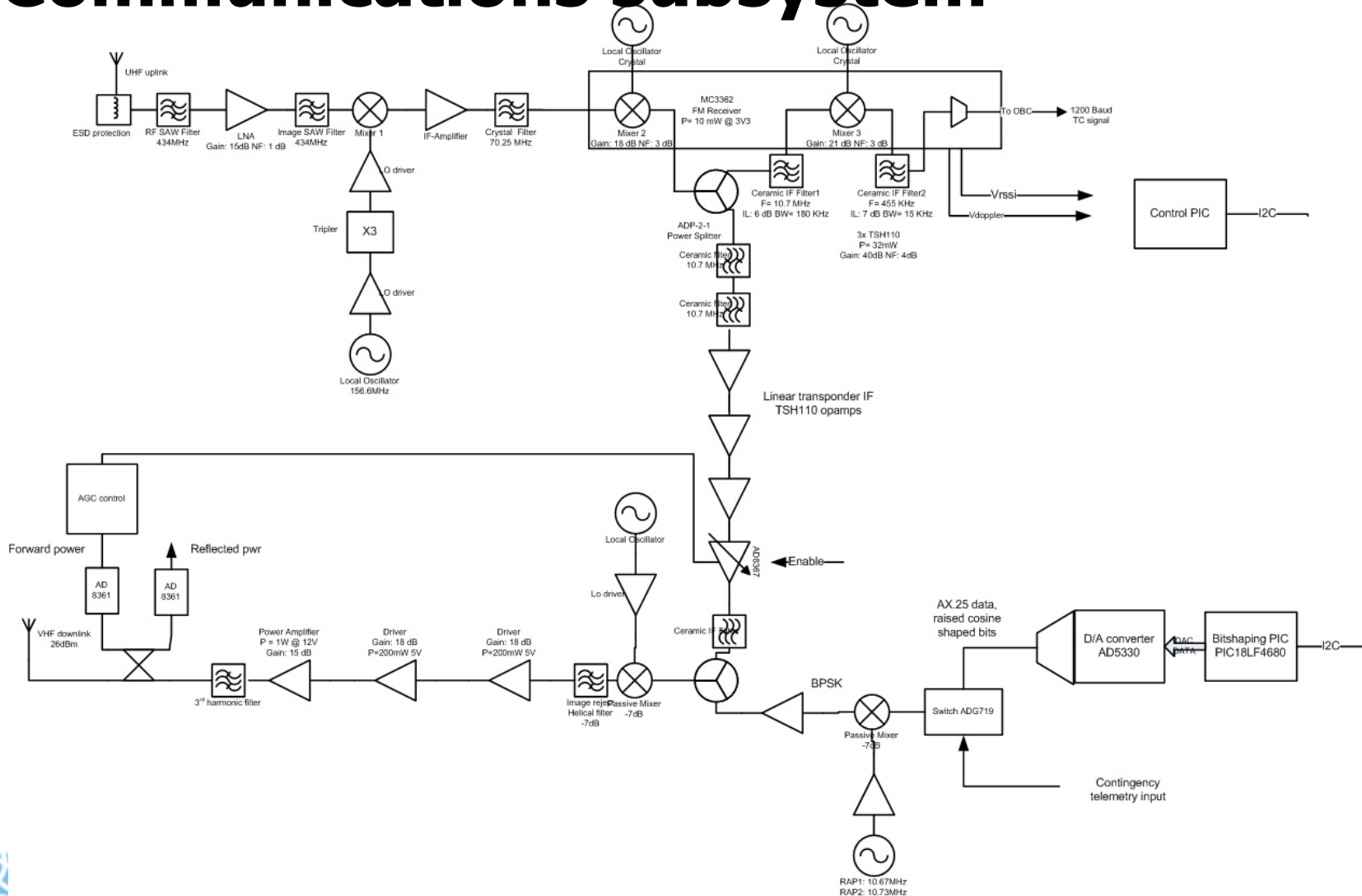
Coordinated by the International Amateur Radio Union (IARU)



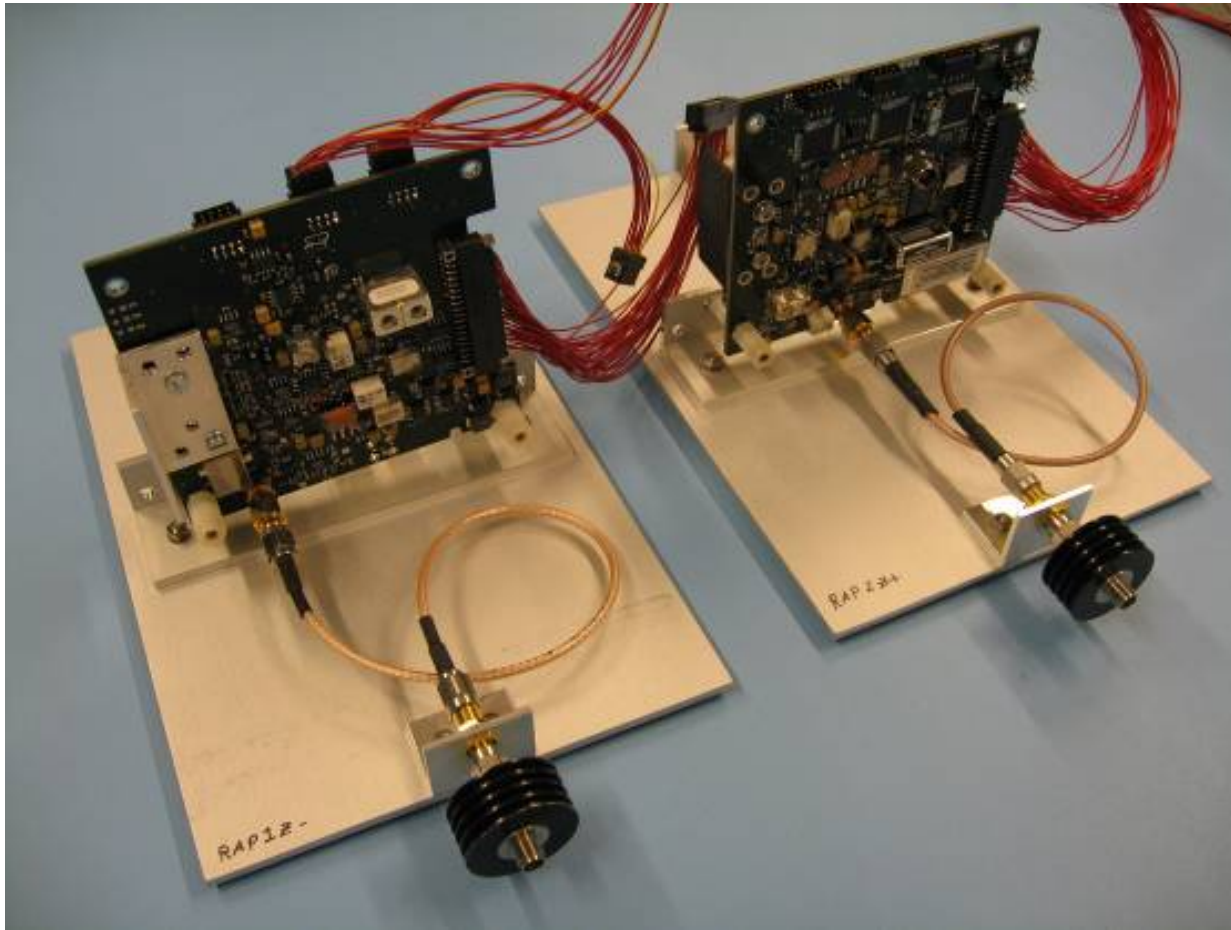
Linear transponder



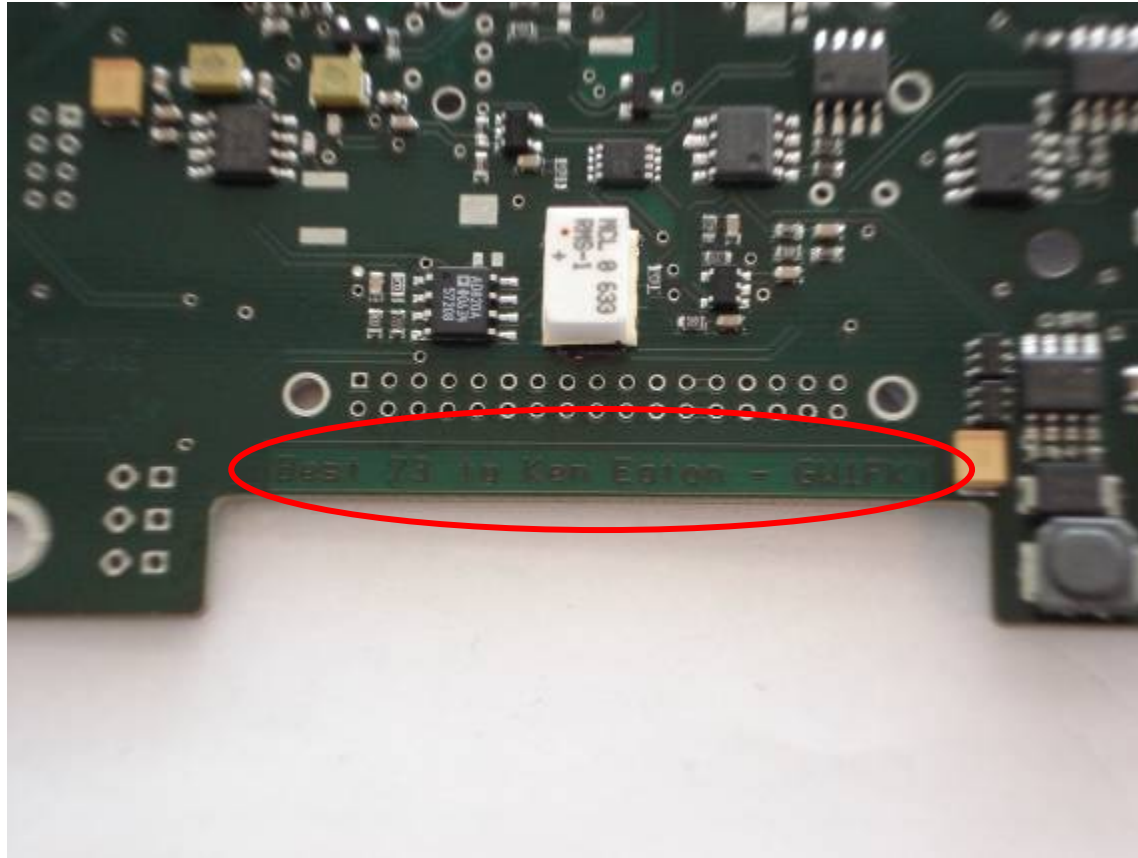
Communications subsystem



Communications subsystem



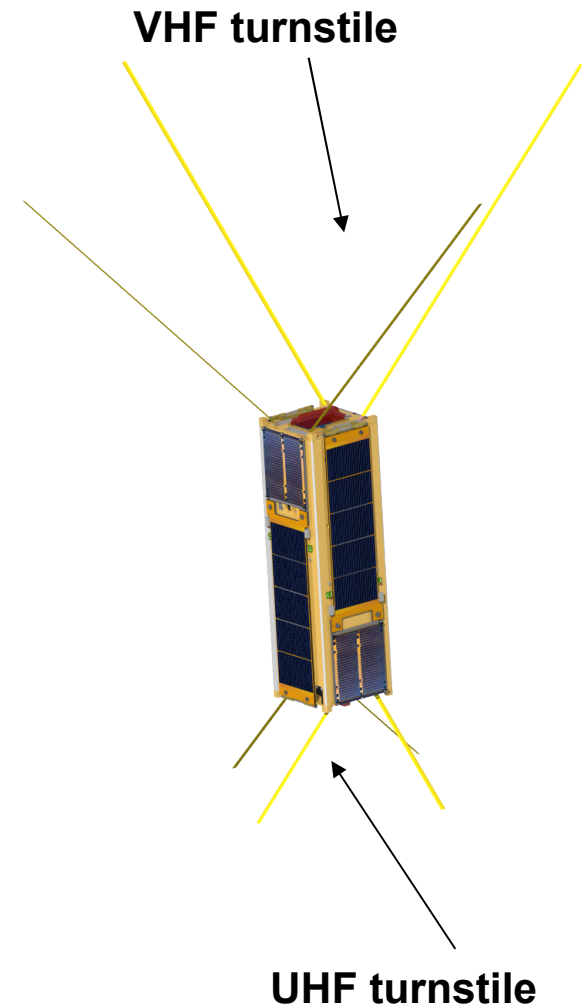
GW1FKY flies aboard



Antenna subsystem

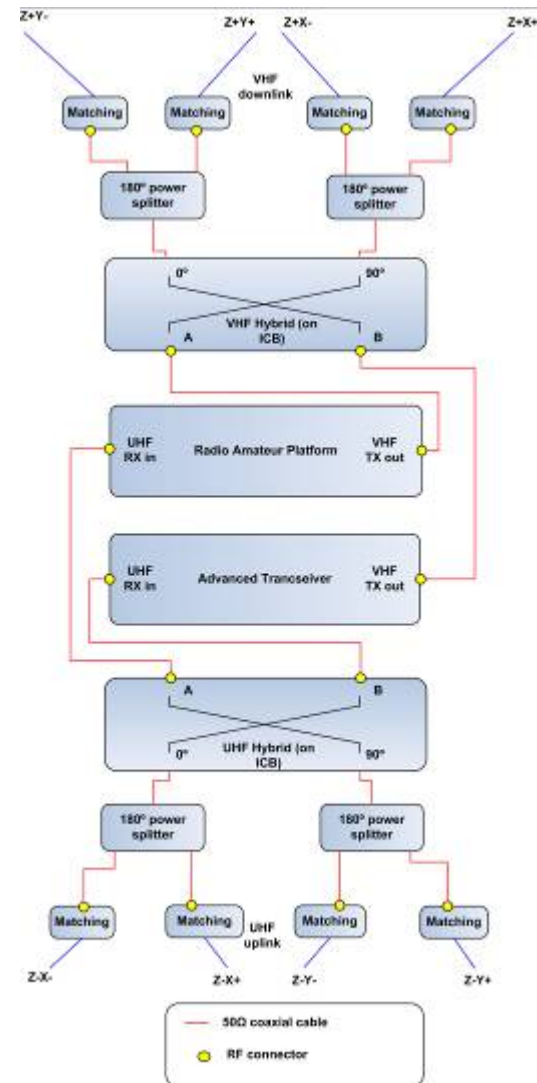
Uplink & downlink: turnstile antenna system

- 4 whips in phase quadrature → pattern and polarization
- VHF: 50cm
- UHF: 18cm
- Phasing harness to achieve phase relationship
- 6mm tape measure antenna whips
- Modular Antenna Boxes

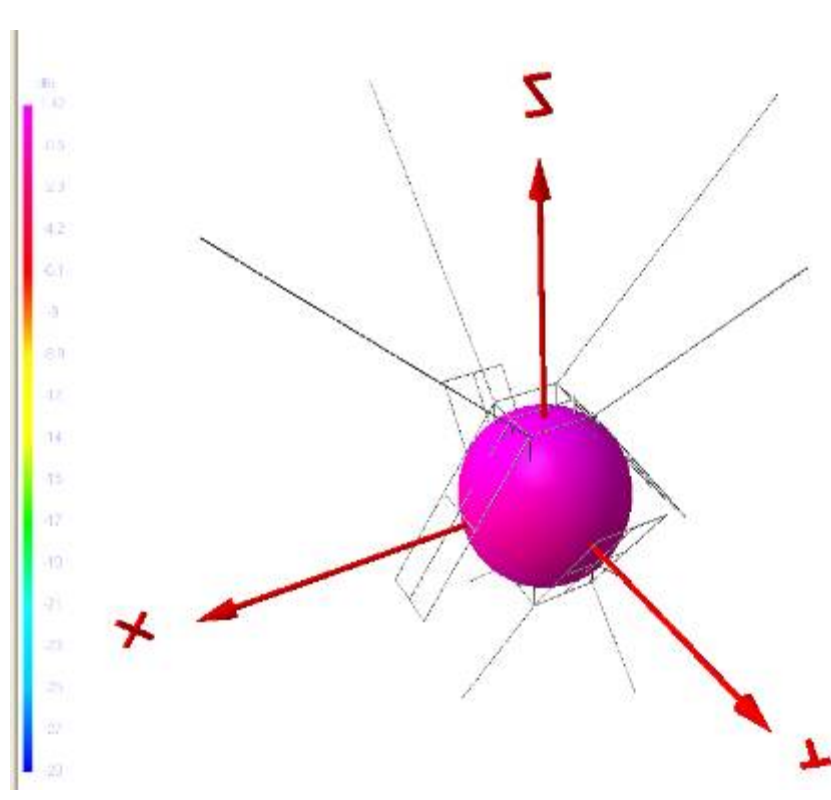


Phasing circuit

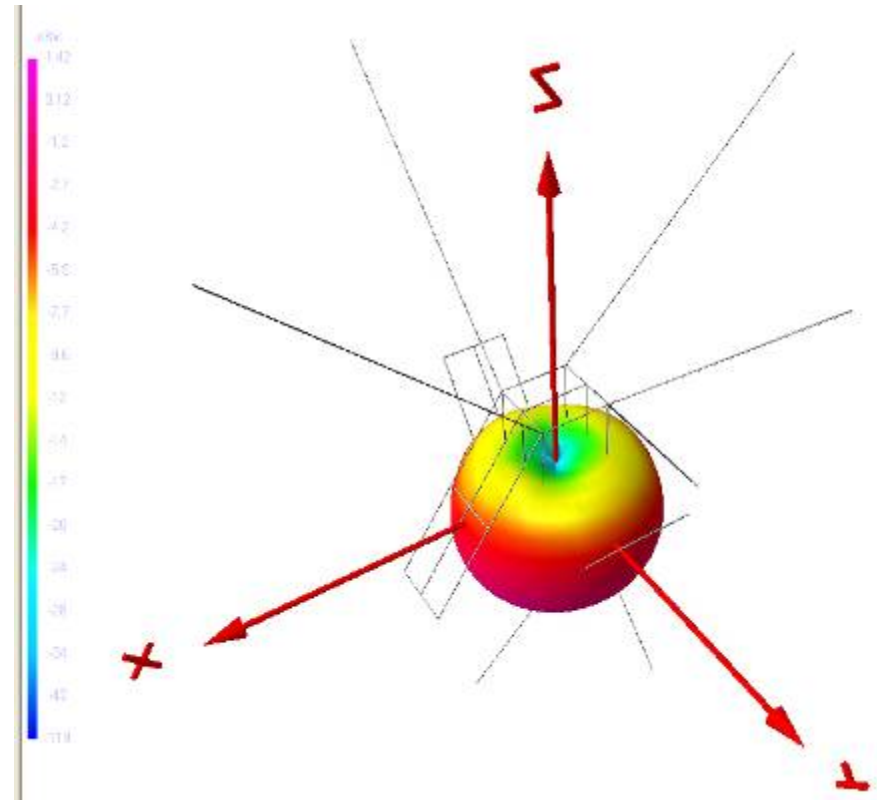
- Quadrature hybrids to split and combine RF from two transceivers
- 180° power splitters to achieve final phase relationship
- Provide ESD discharge path
- Provide isolation
- Low pass filtering
- Provide progressive phase shift per antenna



NEC simulation results



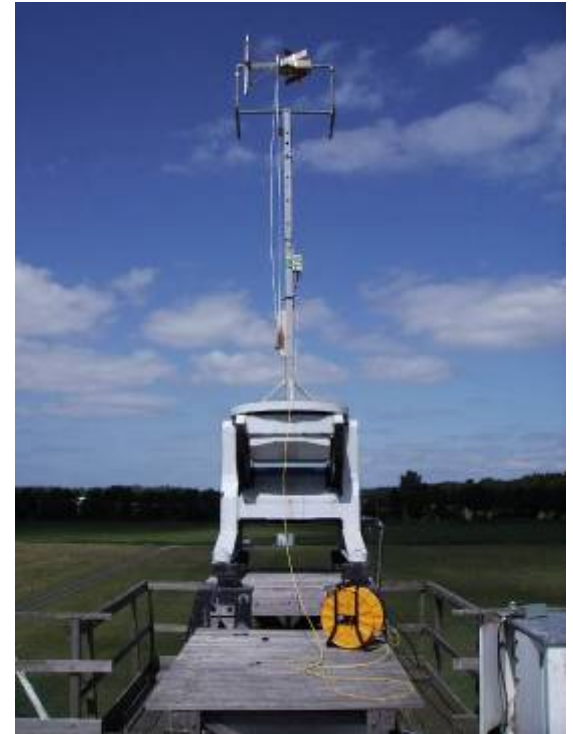
Total gain



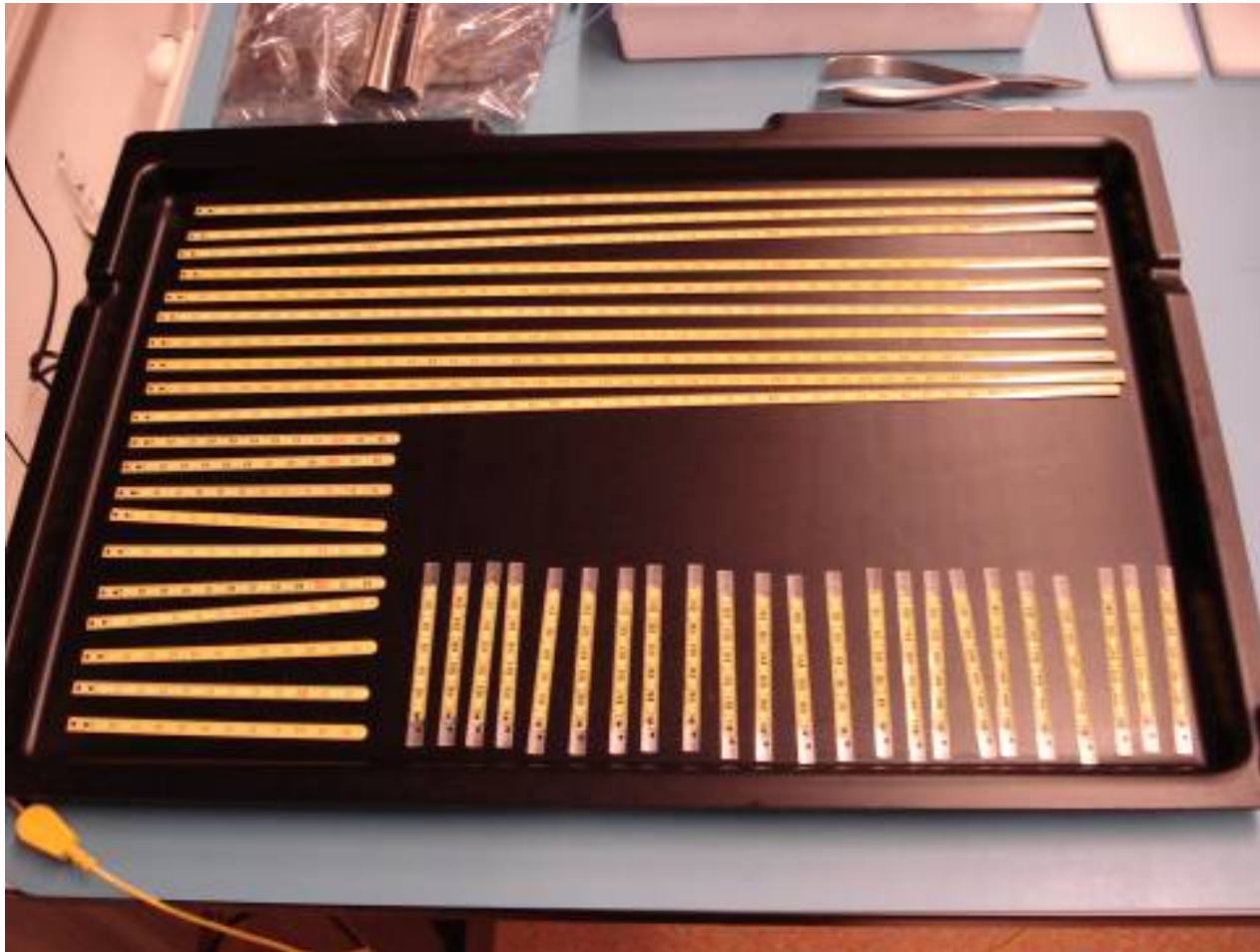
RHCP gain

Antenna testing

- NLR Far Field range
- Verification of VHF / UHF radiation pattern
- Radiation pattern in case of deployment failure



Flight VHF / UHF antennas



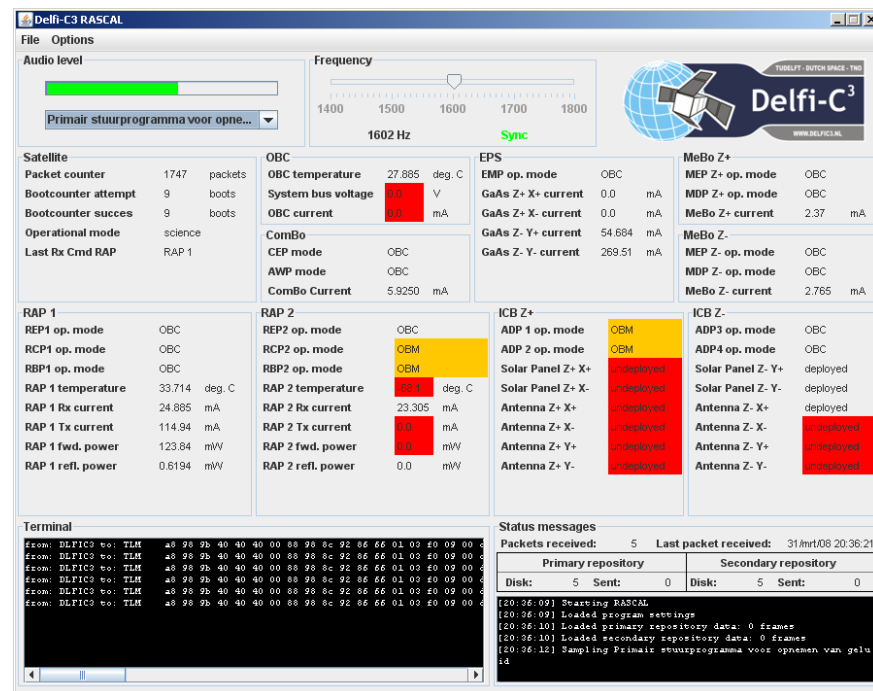
Modular Antenna Boxes





Ground Segment & Data Collection

- Command stations in Delft and Eindhoven (PI4TUE)
- Distributed ground station network
 - Radio amateurs worldwide
 - Universities worldwide
 - Modest setup equipment
 - Software will be made available
- Soundcard software
- RASCAL
 - Displays data realtime
 - Packet storage
 - Website with statistics
 - (amateur competition)
 - Payload data processing
 - Attitude reconstruction
- Satellite status reconstruction / verification
 - Data delivery to customer



Minimum required equipment

- Azimuth/ Elevation rotor
- VHF linearly polarized yagi antenna (preferably 5 elements or more) or
- VHF circularly polarized yagi antenna (preferably 5 elements or more, and with polarization switching)
or
- Omnidirectional antenna (e.g. turnstile, eggbeater) → approx 10dB worse performance, so worst case link margin is 0dB for same BER of 10⁻⁵
- VHF SSB Transceiver or Receiver
- PSK modem and TNC (only required when no soundcard is available)
- Personal Computer running MixW & RASCAL and internet connection



RASCAL

Delfi-C3 RASCAL

File Options

Audio level:

Frequency: 1602 Hz Sync

TUDELFT - DUTCH SPACE - TNO
Delfi-C³
WWW.DELFIC3.NL

Satellite		OBC		EPS		MeBo Z+	
Packet counter	1747 packets	OBC temperature	27.885 deg. C	EMP op. mode	OBC	MEP Z+ op. mode	OBC
Bootcounter attempt	9 boots	System bus voltage	0.0 V	GaAs Z+ X+ current	0.0 mA	MDP Z+ op. mode	OBC
Bootcounter succes	9 boots	OBC current	0.0 mA	GaAs Z+ X- current	0.0 mA	MeBo Z+ current	2.37 mA
Operational mode	science	ComBo		GaAs Z- Y+ current	54.684 mA	MeBo Z-	
Last Rx Cmd RAP	RAP 1	CEP mode	OBC	GaAs Z- Y- current	269.51 mA	MEP Z- op. mode	OBC
		AWP mode	OBC			MDP Z- op. mode	OBC
		ComBo Current	5.9250 mA			MeBo Z- current	2.765 mA

RAP 1		RAP 2		ICB Z+		ICB Z-	
REP1 op. mode	OBC	REP2 op. mode	OBC	ADP 1 op. mode	OBM	ADP3 op. mode	OBC
RCP1 op. mode	OBC	RCP2 op. mode	OBM	ADP 2 op. mode	OBM	ADP4 op. mode	OBC
RBP1 op. mode	OBC	RBP2 op. mode	OBM	Solar Panel Z+ X+	undeployed	Solar Panel Z- Y+	deployed
RAP 1 temperature	33.714 deg. C	RAP 2 temperature	-68.1 deg. C	Solar Panel Z+ X-	undeployed	Solar Panel Z- Y-	deployed
RAP 1 Rx current	24.885 mA	RAP 2 Rx current	23.305 mA	Antenna Z+ X+	undeployed	Antenna Z- X+	deployed
RAP 1 Tx current	114.94 mA	RAP 2 Tx current	0.0 mA	Antenna Z+ X-	undeployed	Antenna Z- X-	undeployed
RAP 1 fwd. power	123.84 mW	RAP 2 fwd. power	0.0 mW	Antenna Z+ Y+	undeployed	Antenna Z- Y+	undeployed
RAP 1 refl. power	0.6194 mW	RAP 2 refl. power	0.0 mW	Antenna Z+ Y-	undeployed	Antenna Z- Y-	undeployed

Terminal

```

from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d
from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d
from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d
from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d
from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d
from: DLFIC3 to: TLM  a8 98 9b 40 40 40 00 88 98 8c 92 86 66 01 03 f0 09 00 d

```

Status messages

Packets received: 5 Last packet received: 31/mrt/08 20:36:21

Primary repository		Secondary repository	
Disk:	5 Sent: 0	Disk:	5 Sent: 0

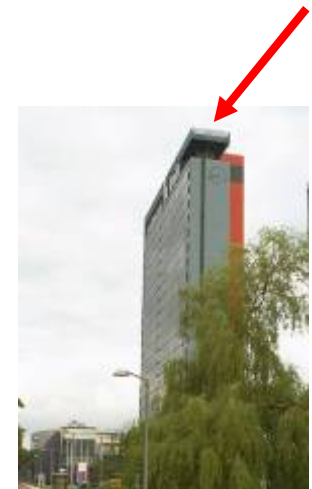
```

[20:36:09] Starting RASCAL
[20:36:09] Loaded program settings
[20:36:10] Loaded primary repository data: 0 frames
[20:36:10] Loaded secondary repository data: 0 frames
[20:36:12] Sampling Primair stuurprogramma voor opnemen van gelu
id

```

Ground Station

- Fully automated
- Tracking yagi antennas
- VHF / UHF / S-Band
- Backup power
- Tracking and decoding telemetry from LEO satellites
- Remotely controllable



Soundclips

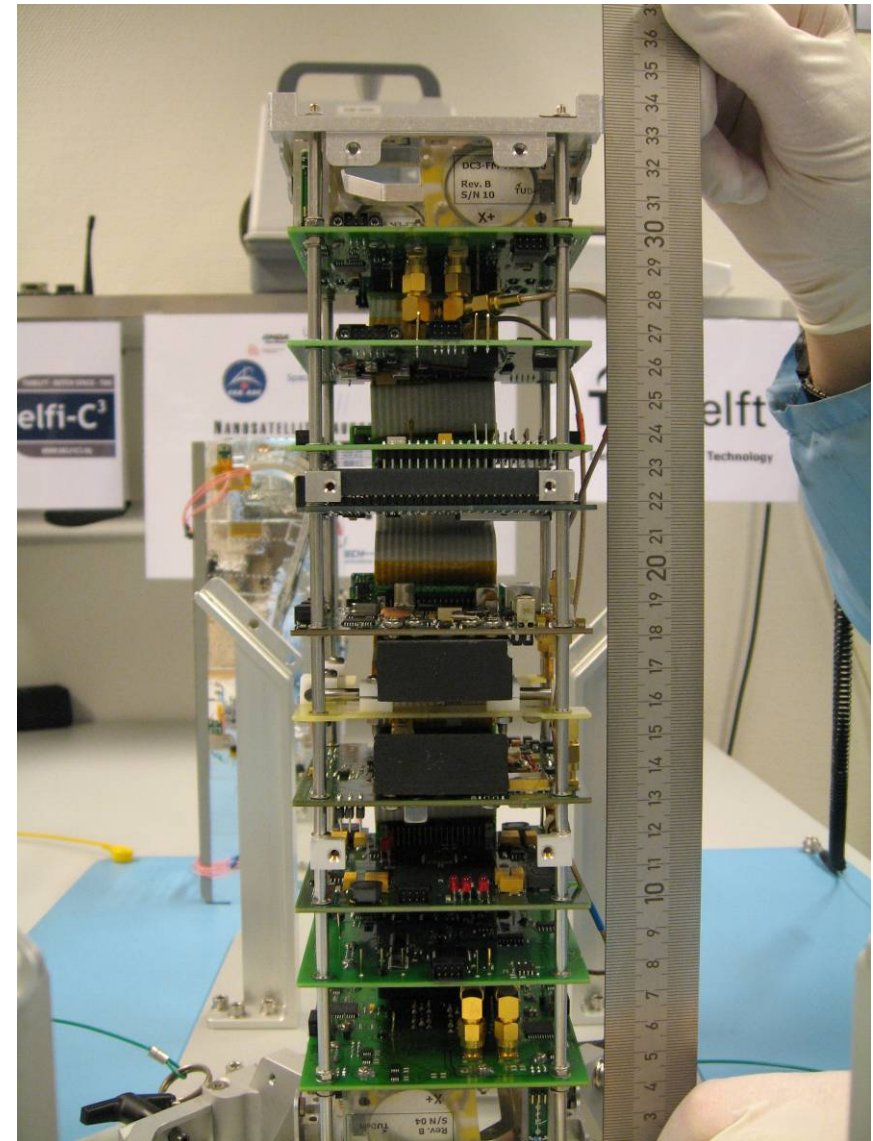
Beacon

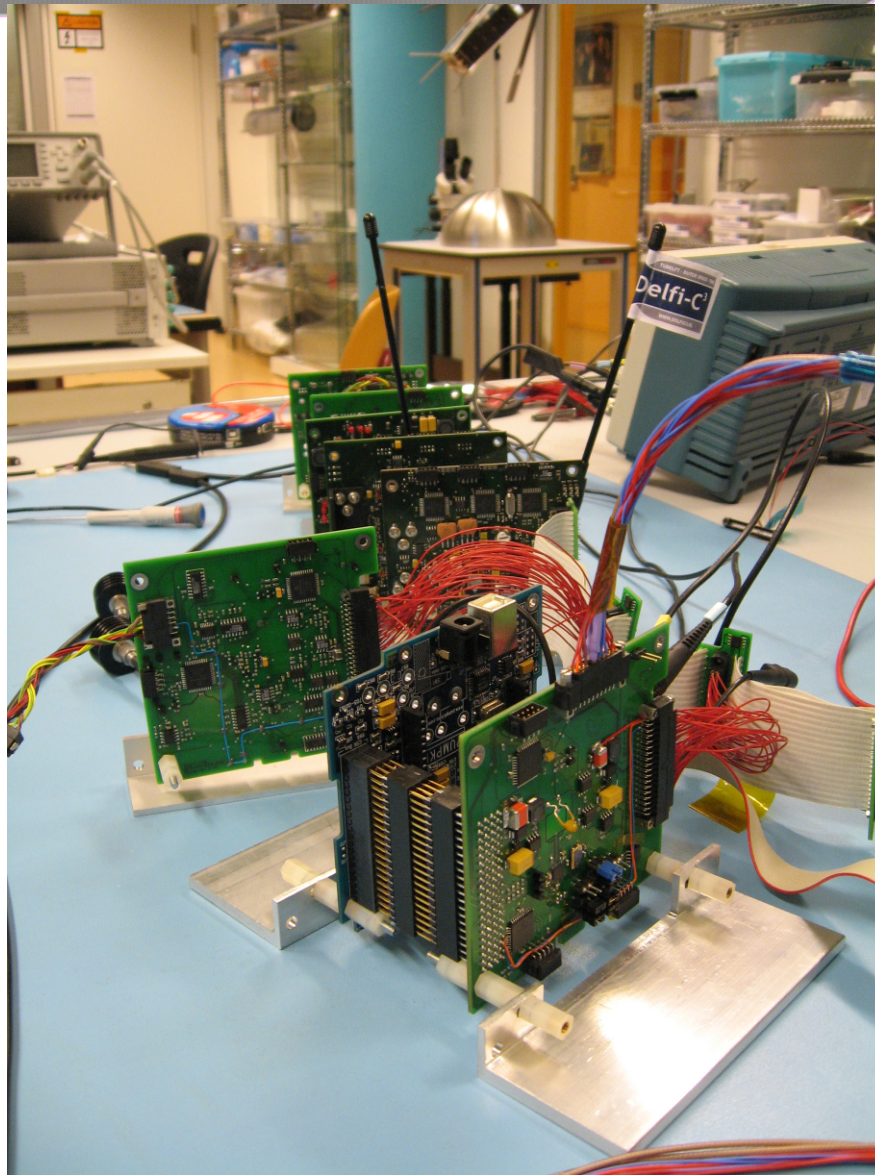


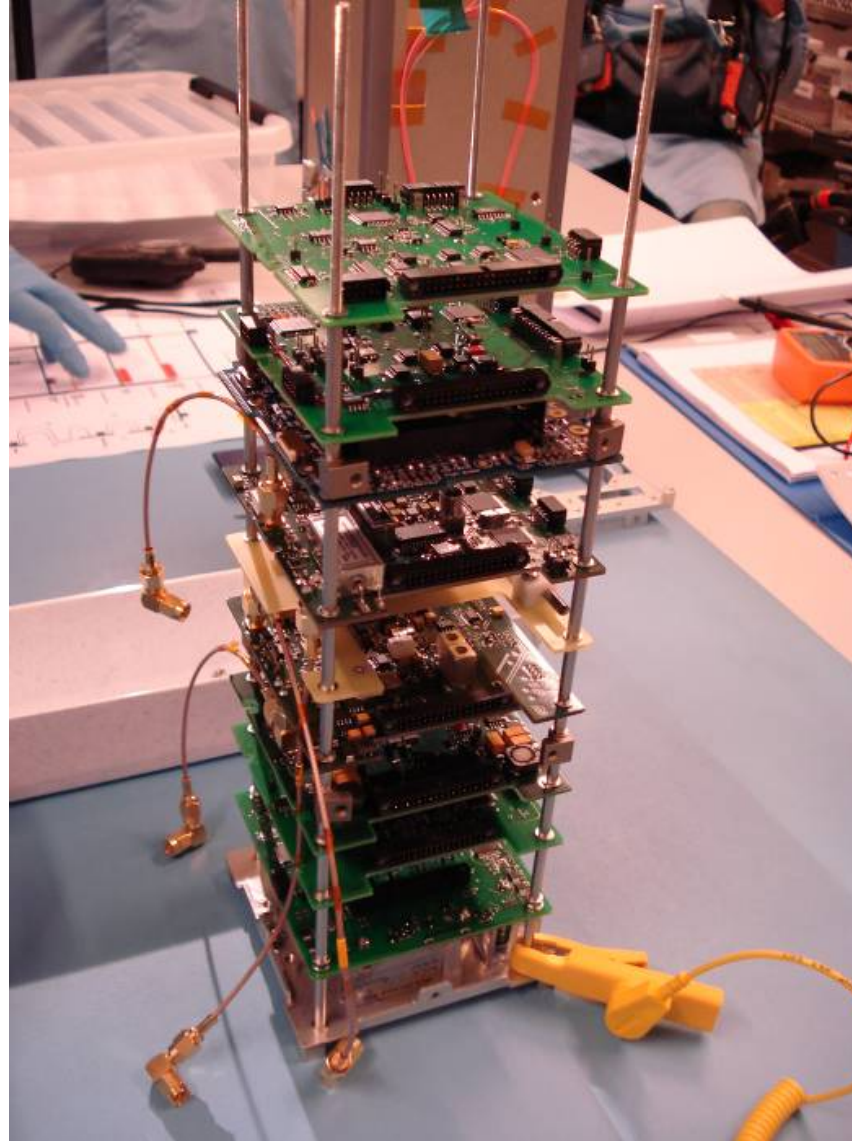
First test link



Delfi-C3 flight stack

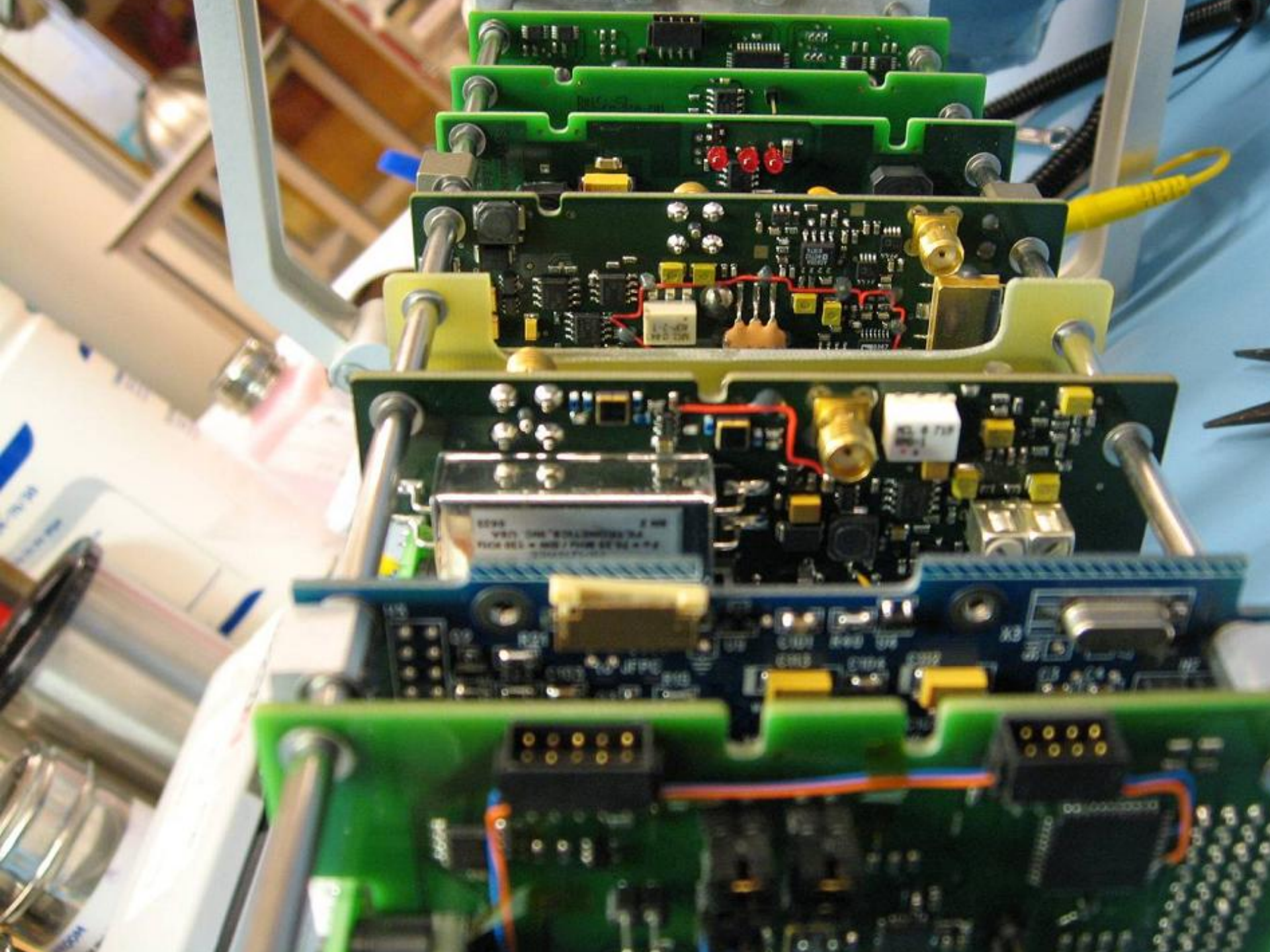


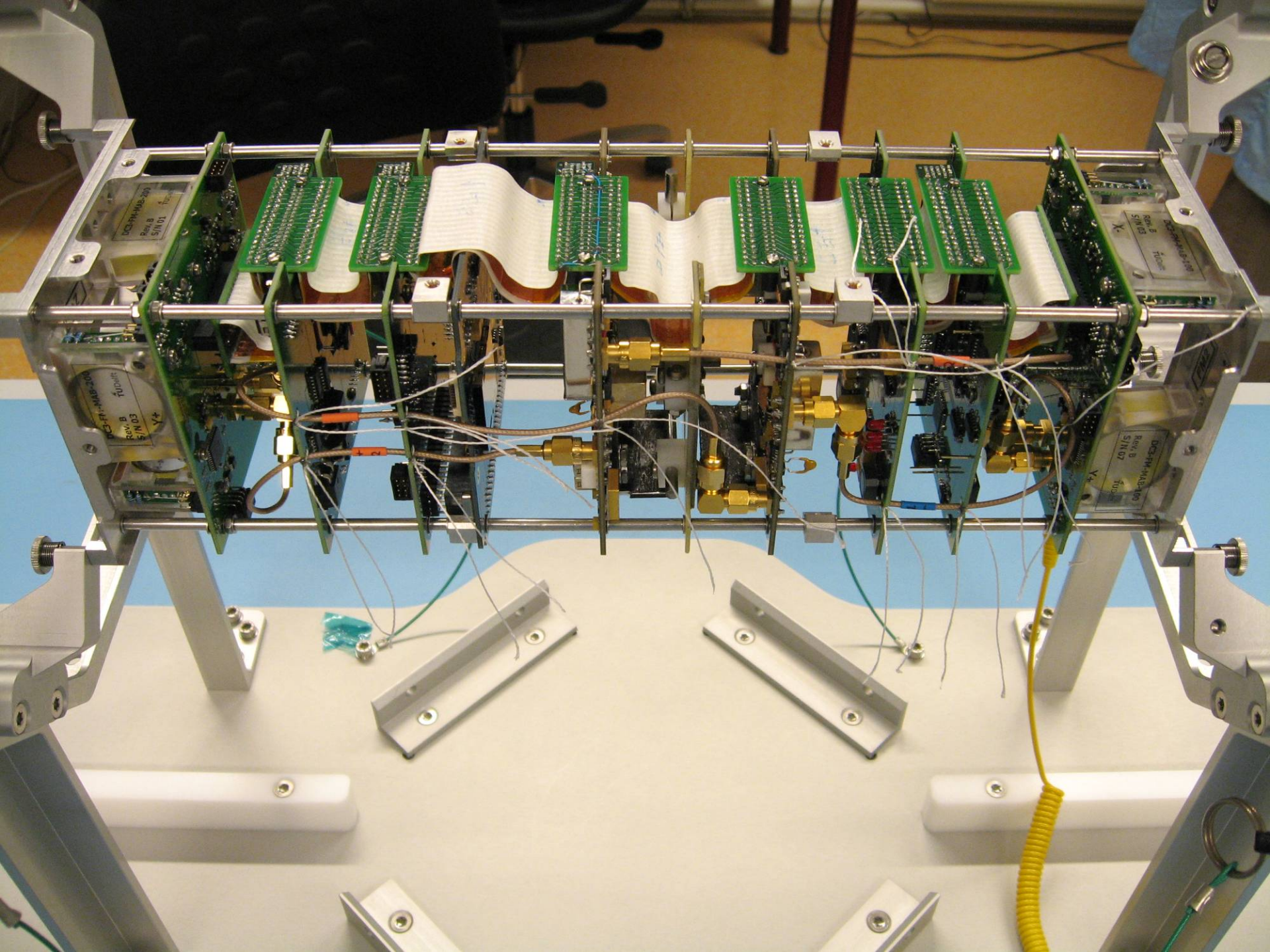




VERON Friese Wouden



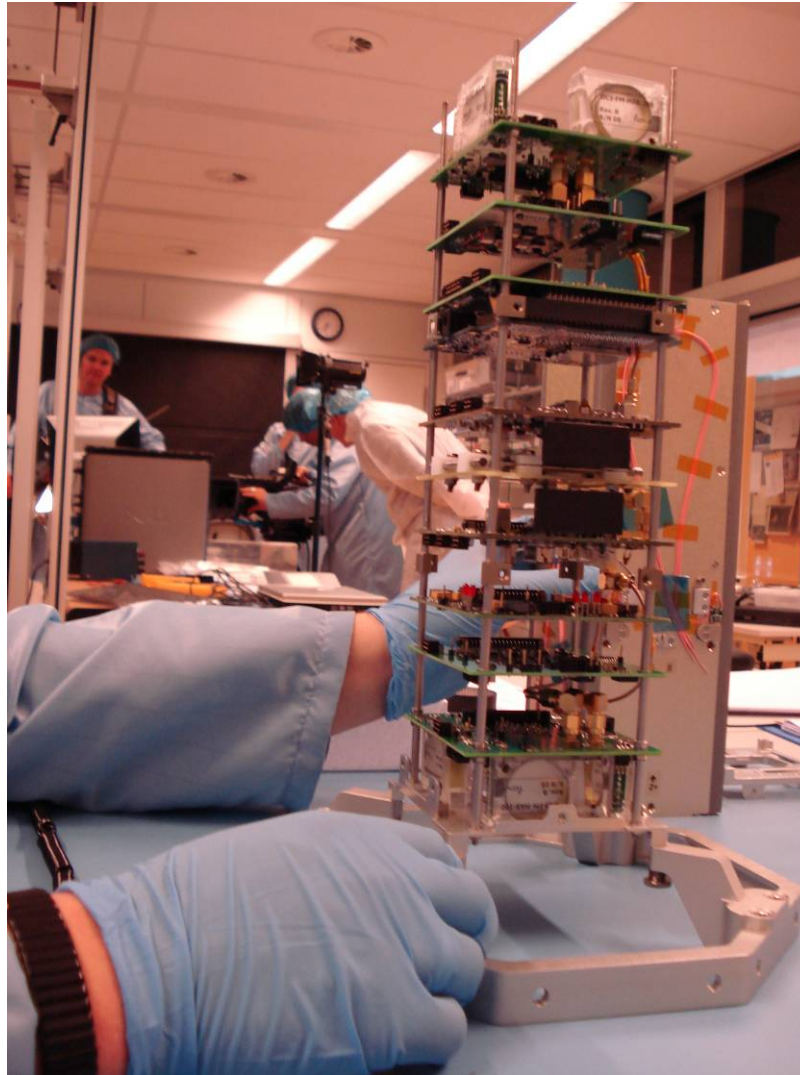




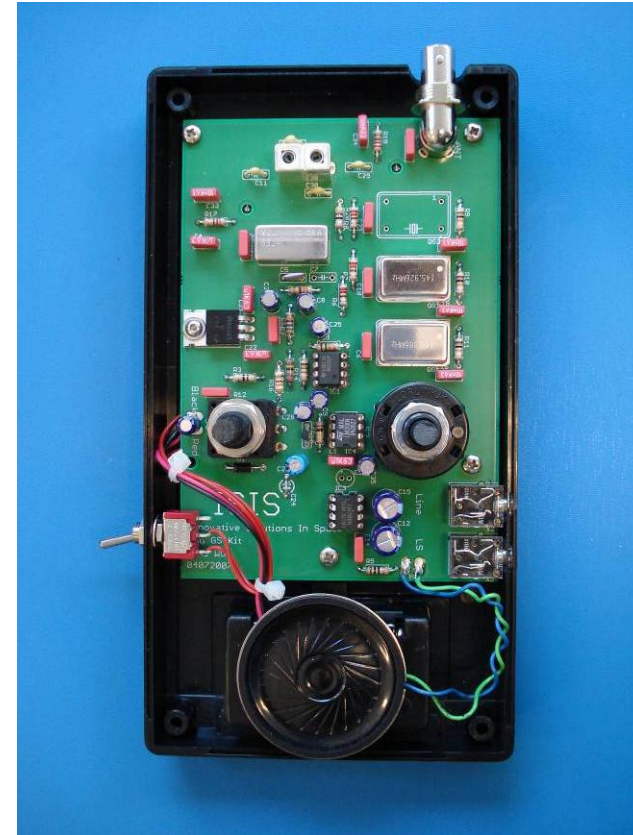
Klokhuis (dutch TV)



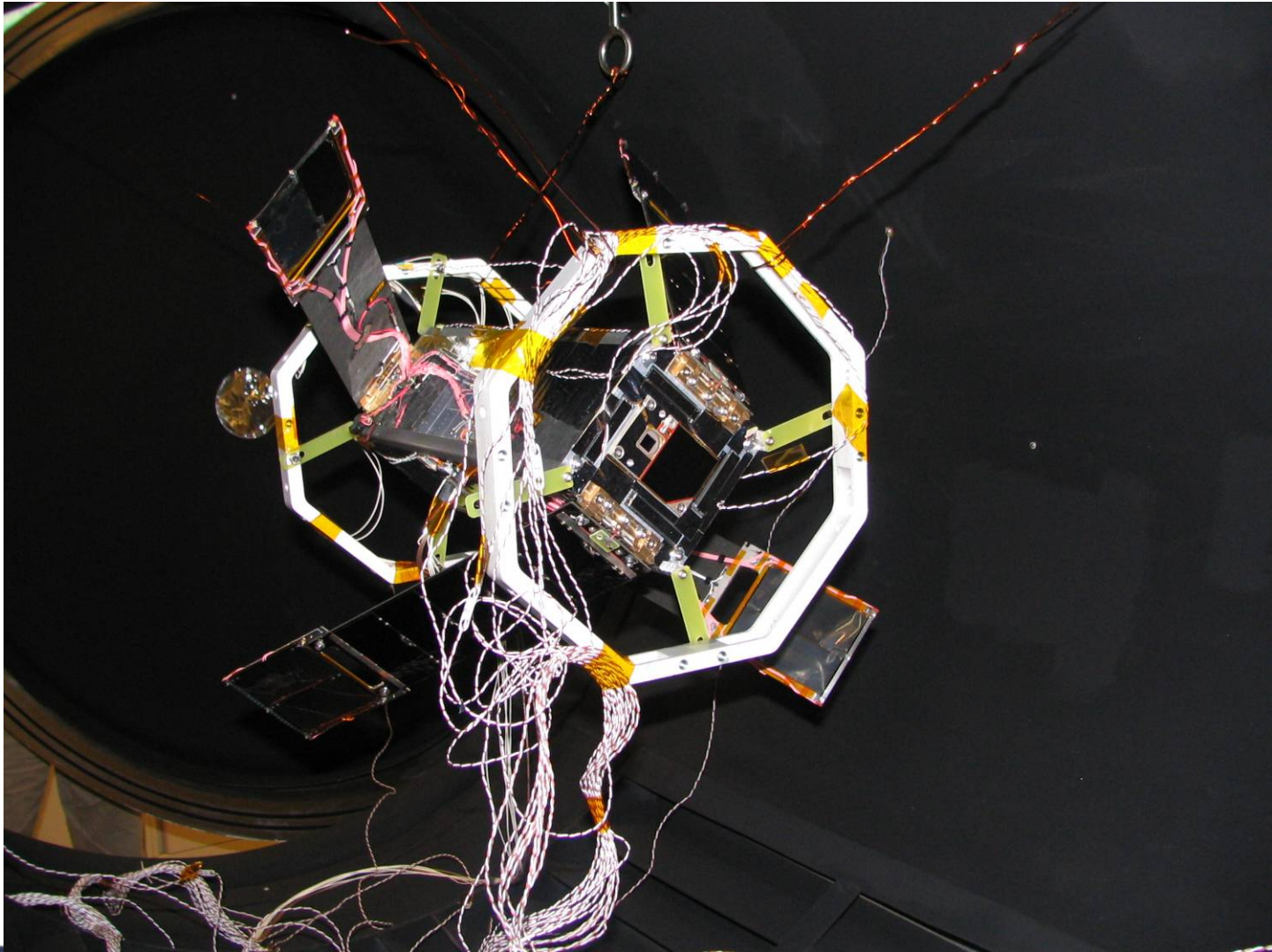
Klokhuis



High school receiver kit



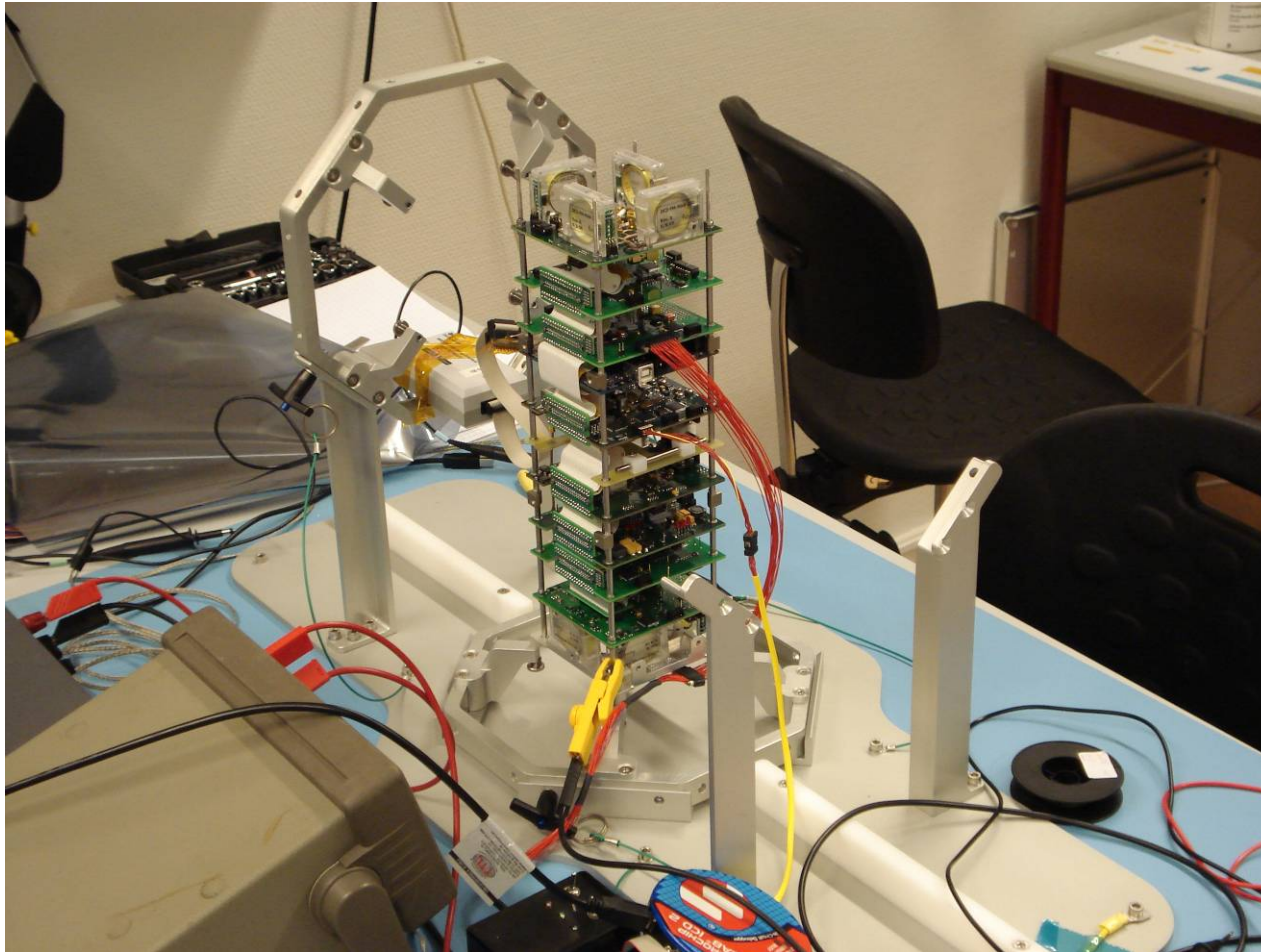
TVAC Test



VERON Friese Wouden



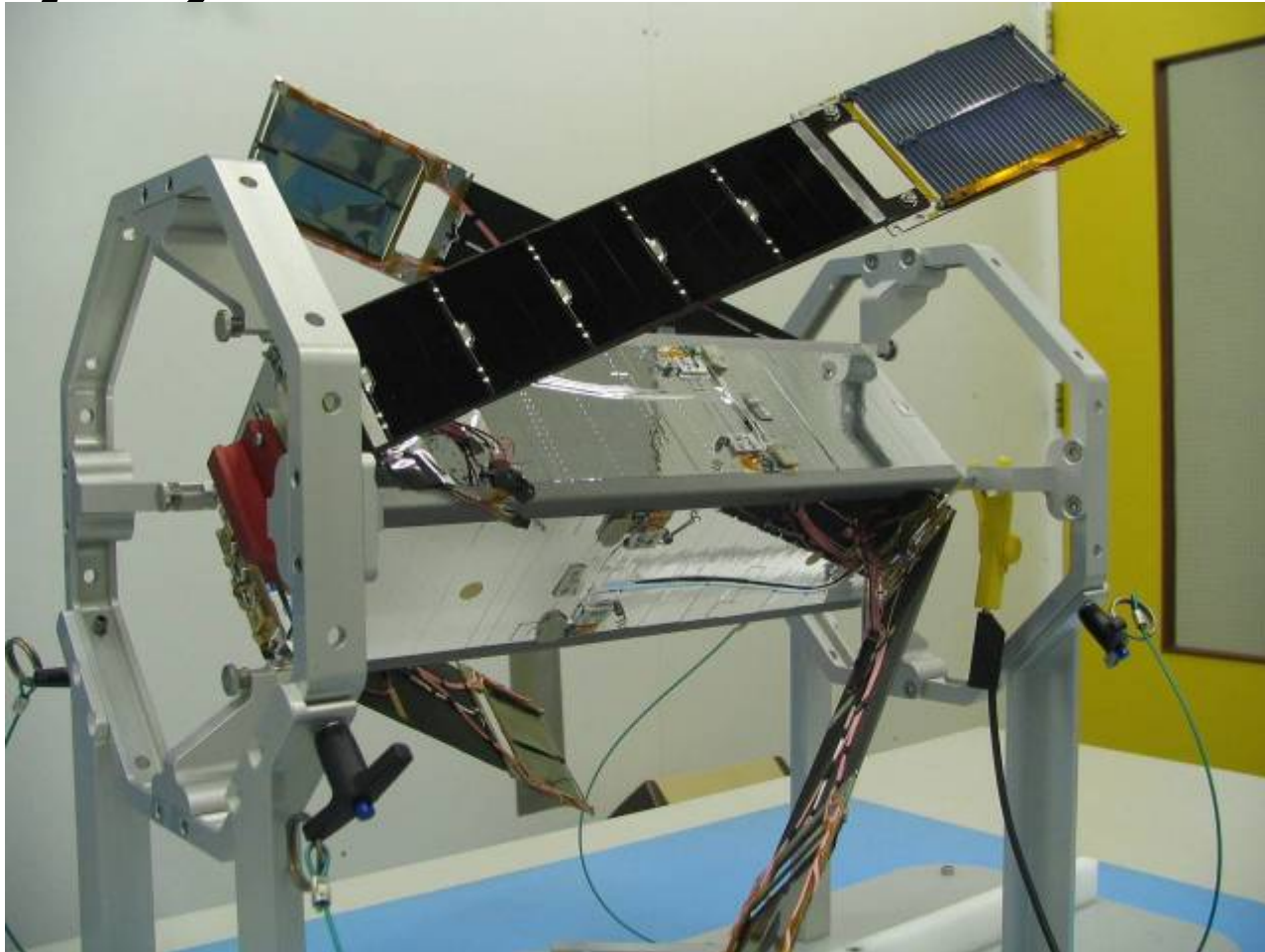
Testing



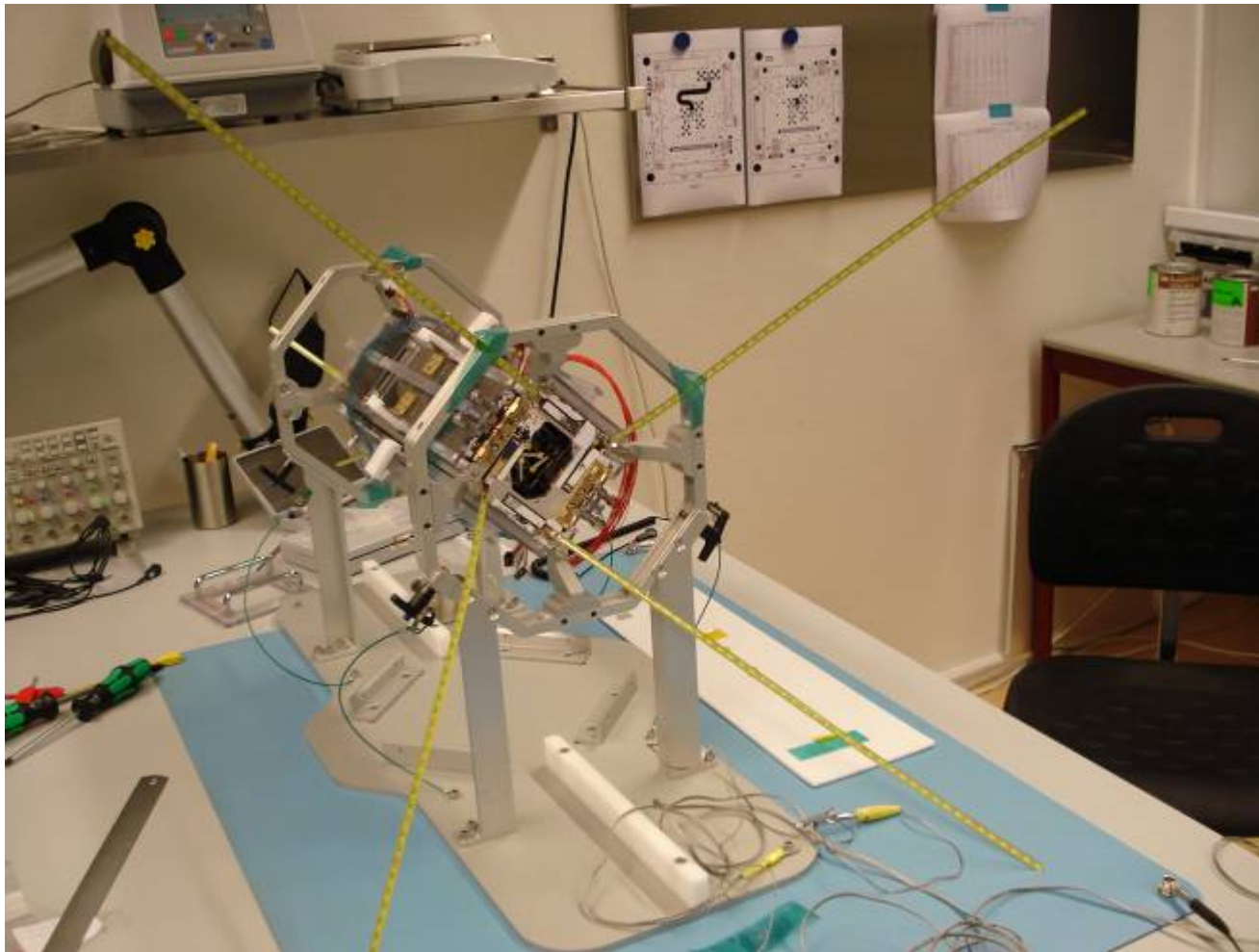
Vibe Test

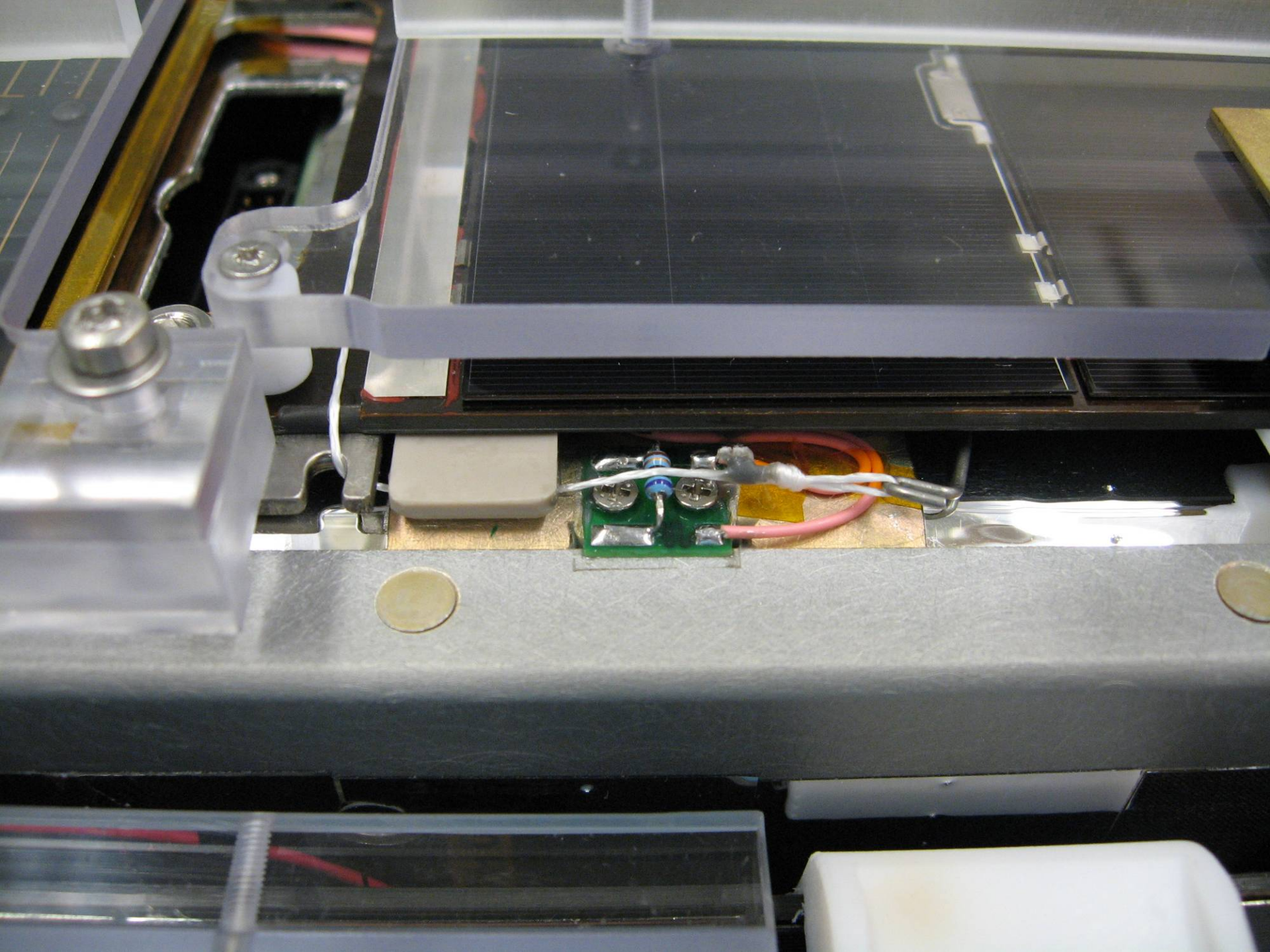


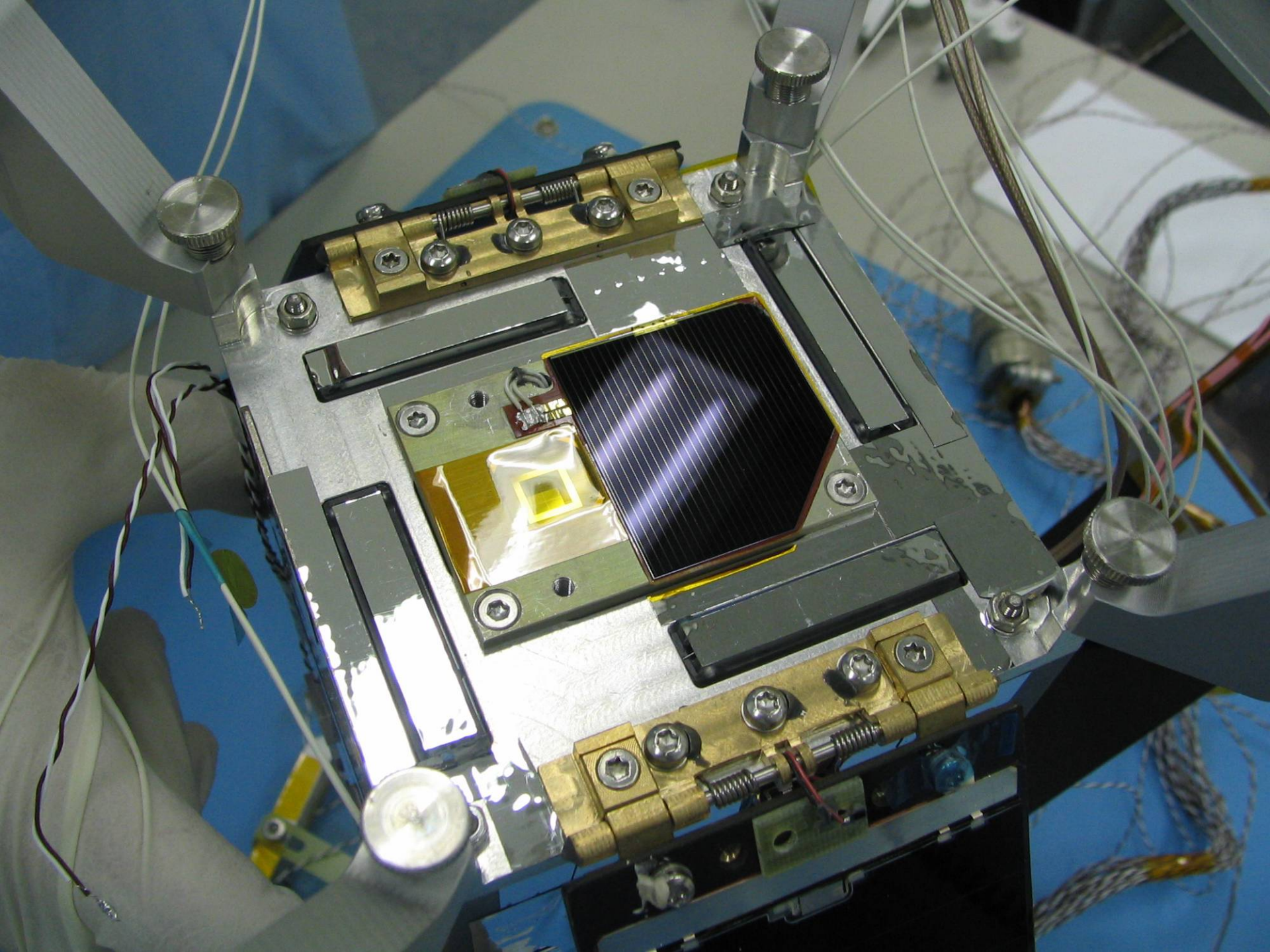
Flight configuration (antennas not deployed)



Deployed antennas









Static safe working?



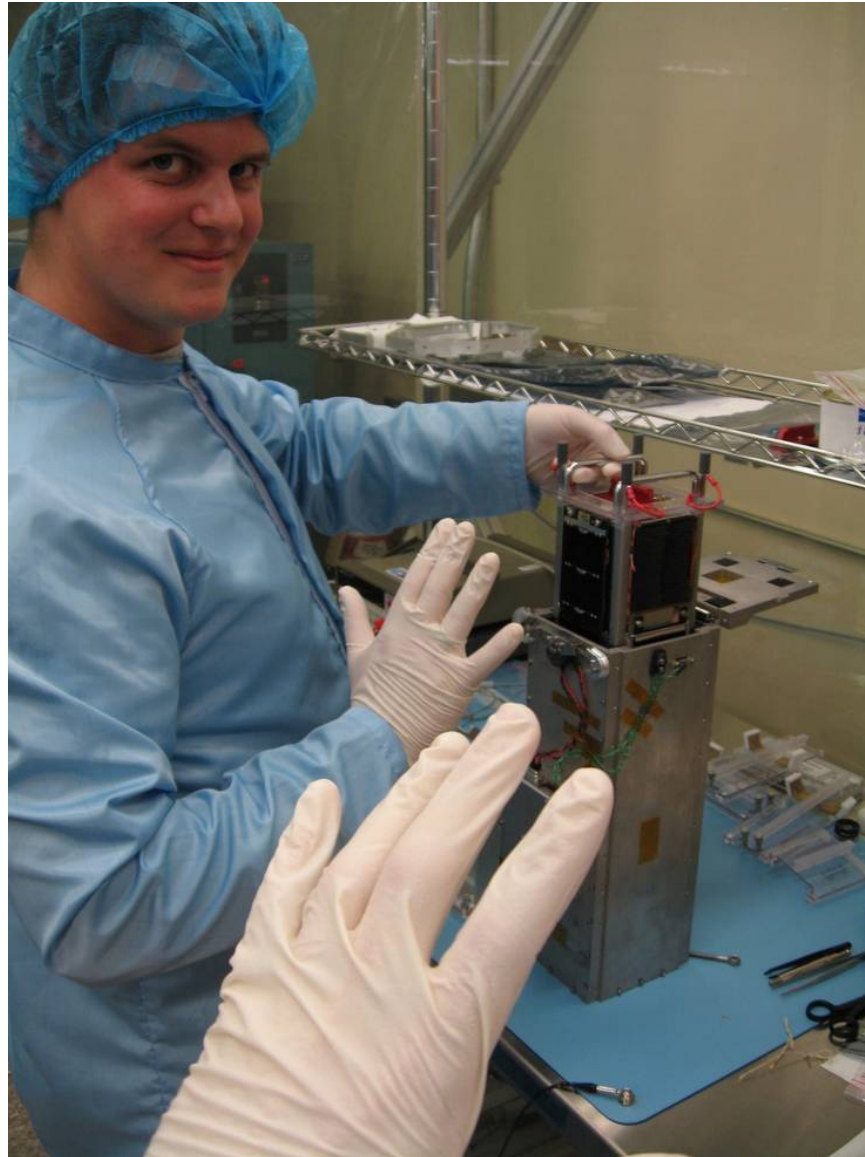
PE4WJ

DELFI-C³

PA3WEG



Bye Bye



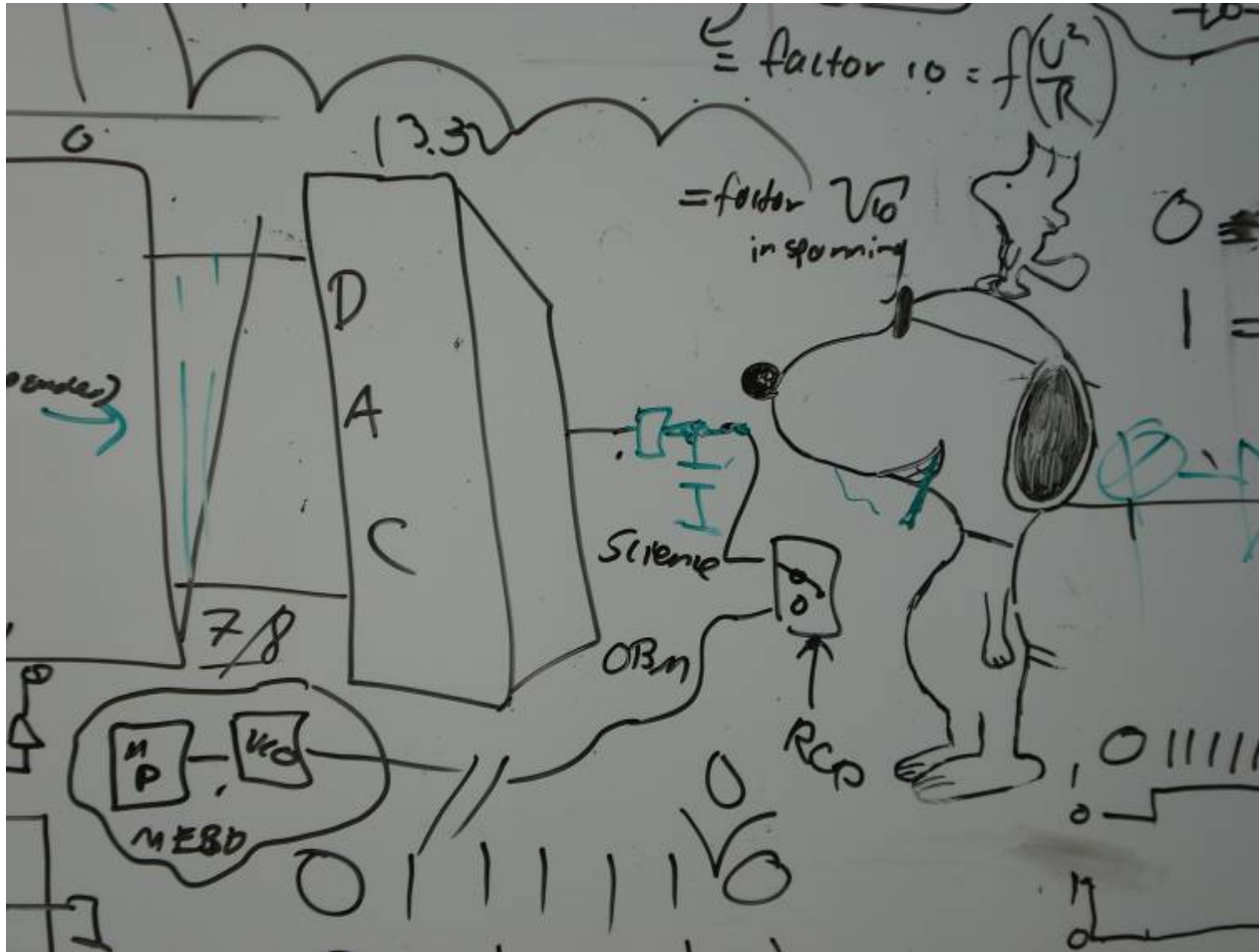
VERON Friese Wouden



Space Art



How to design a satellite...



Lessons Learnt

- Start building prototypes early!
- In RF circuits, power matching is not always necessary
- Opamps make nice RF/IF amplifiers
- Do not choose 0000 as bus reset command
- Parasitic oscillations can popup just about anywhere
- I2C repeaters / pull down by PICs
- Use ground lines between I2C data lines (**Doh!**)
- Reserve spare pins on connectors, you will need them!
- Use 0-ohm resistors to connect subcircuits
- **And...**



...Satellites are indeed, entirely constructed out of pizzas



Latest photos

- <http://www.delfic3.nl/photoblog/>



QRZ?



www.delfic3.nl
wouterw@delfic3.nl
info@delfic3.nl